



# Unmanned Aerial Vehicle (UAV) Cargo System

For Integrated Communications  
Navigation and Surveillance  
(ICNS) Conference

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Systems Engineering and Operational Research Department



# Outline

- Background
- Problem & Mission Statement
- System **Requirements** vs. **Technology** Capability
- Describe the **Concept of Operations**
- Describe the **Avionics Architecture**
  - **Components**
  - **Relationship between components**
- Describe the results of **Reliability Analysis**
- Summary of **Business case**



# Background

- Driven by economic demand that is not currently being satisfied
  - Cargo demand increasing 15%/year internationally
  - 7%/year domestically

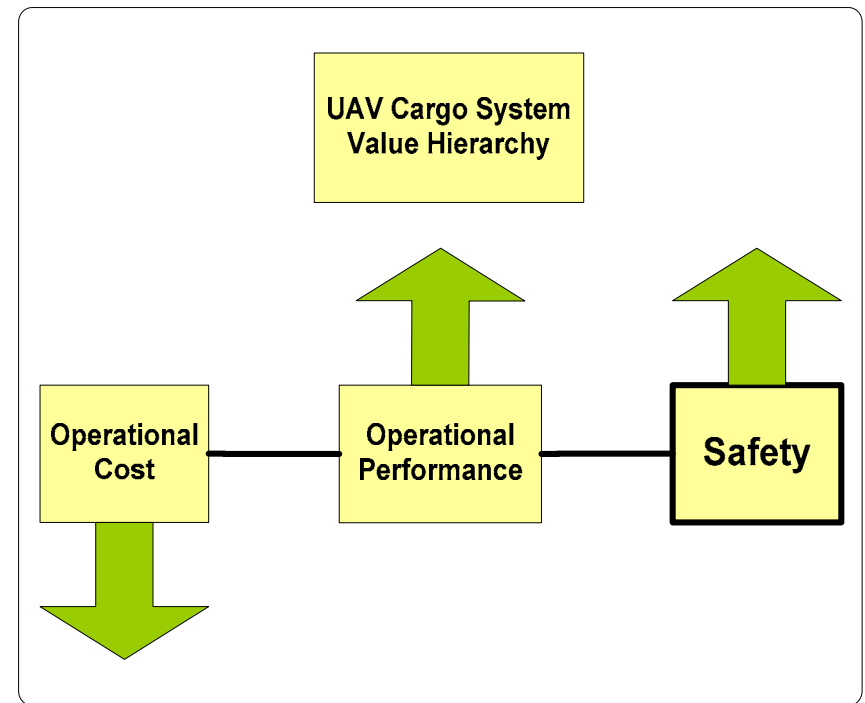
Reference: Rediess, Herman A. "Airport-Independent Uninhabited Air Vehicle Cargo System Concept"

Source: 1995-2002; U.S. Air Carriers, Form 41, U. S. Department of Transportation., [www.dot.gov/strategic-plan](http://www.dot.gov/strategic-plan)



# Problem Statement

- Develop a preliminary design of a UAV Cargo System
- Simulate and analyze the safety and feasibility of the system performance
- Analyze costs and benefits for investment decision
- Ultimately to be certified by FAA
- Determine the feasibility of the business case



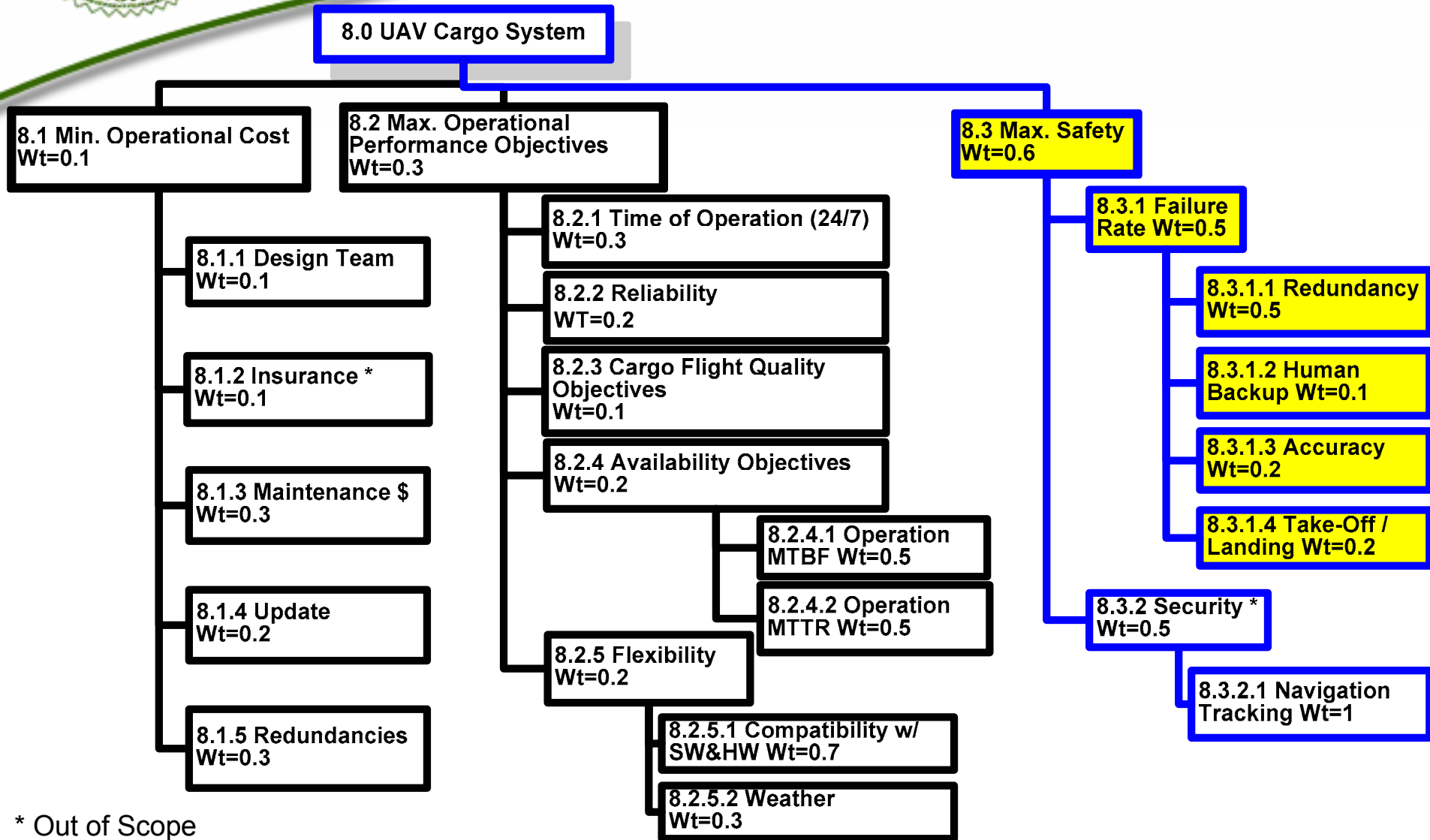


# Mission Statement

- Design an Unmanned Aerial Vehicle (UAV) Cargo System
  - **8 hours Just In Time (JIT) delivery for business case**
  - **Provide new level of cargo service 24/7 for light weight/high value cargo**
  - Meet the FAA Safety Standard
  - Automated
  - Efficient
  - Cost effective
- Public Airport-independent
- Provide an all weather autonomous landing capability
  - Achieve fully coupled precision approach meaning it will autonomously land with flight guidance and control system (comparable level of performance like CAT III B)



# Value Hierarchy Overview





# **Requirements vs. Technology Capability**



# Requirement

## **“What”**

- System shall provide new level of service
  - Just In Time (JIT) delivery within 8 hours

## **“How”**

- Air Transportation
  - Operate directly from and to Industrial Park (IP)
  - Use existing airplane and avionics





# View Graph Example

**Class A Airspace**

**> 18,000 Ft.**

**< 18,000 Ft.**

**UAV Cargo Planes**



**Cessna Grand Caravan**

**Industrial Park 1**



**Industrial Park 2**





# Current Cessna Caravan

- *Cessna Caravan* is already used for cargo delivery service which is certified by FAA
- GPS Navigation
  - TSO- C129 A1
  - Approach Certified
- Multi Function Display
  - Traffic (ADS-B)
  - Terrain
  - Weather (FIS-B)
- Data link Radio
  - UAT





# Requirement

## **“What”**

- Autonomous Takeoff, En-Route, and Landing

## **“How”**

- Avionics (e.g. fully coupled precision approach = CAT III B auto landing)
  - WAAS/GPS
  - Radar Altimeter
  - EGPWS
  - Vision Positioning System



# Requirement

## **“What”**

- See (Senses) and Avoid

## **“How”**

- Redundant and Diverse Collision Avoidance & Surveillance Sensors
- Aircraft Collision Avoidance
  - ADS-B
  - TIS-B
  - TCAS II
- Terrain Collision Avoidance
  - EGPWS
  - Radar Altimeter
  - Vision Positioning System



# Requirement

## **“What”**

- Avoid Severe Convective Weather Conditions Autonomously

## **“How”**

- Weather Avoidance Avionics
  - FIS-B
    - Icing Altitude (Icing Boots)
    - Convective Weather Cells



# Requirement

## **“What”**

- Meet FAA Safety Regulations

## **“How”**

- Class E Airspace
- Fly under IFR
- Monitored by Human Operator in Ground Station
  - Monitor up to 6~7 UAVs and increase as experience gained





# Requirement

## **“What”**

- No Major Ground Infrastructure Changes in IPs

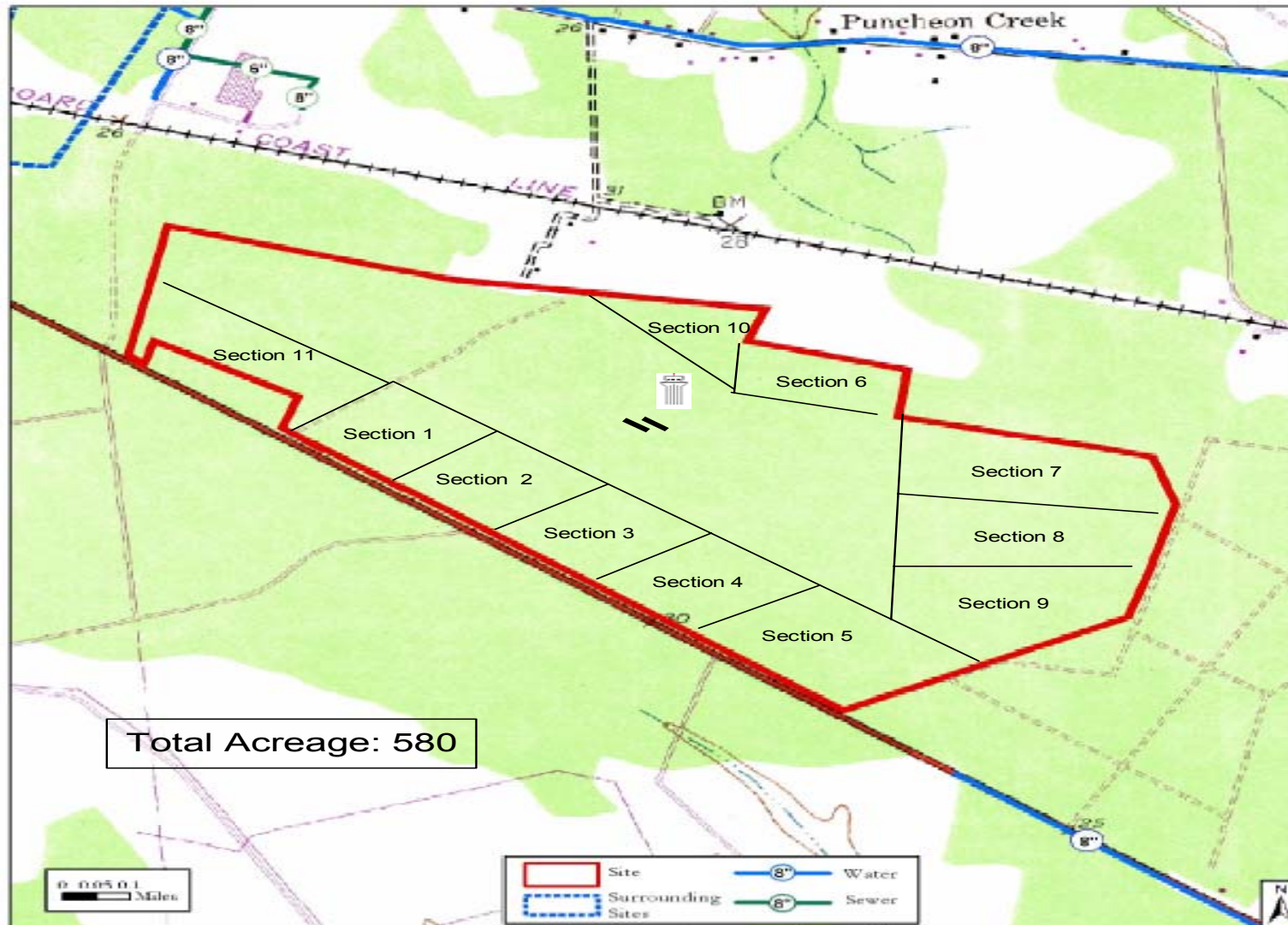
## **“How”**

- Data Links
- ADS-B Ground Station
- TIS-B Ground Station
- FIS-B Ground Station
- IP parking lots for runways



# Runway Space for IP

## Georgetown County Business Park, SC



Reference:

<http://www.teamsc.com/teamscpdfs/siteFliers/844-Topo%20Map.pdf>





# Overall System Assumptions

- Public Airport-independent ground infrastructure and Communication Data Link between ATC-UAVs already in place and operate accurately and reliably
- Sufficiently reliable Landing will ensure the Take-Off portion of the flight



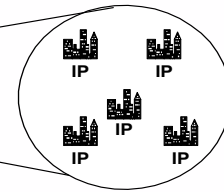
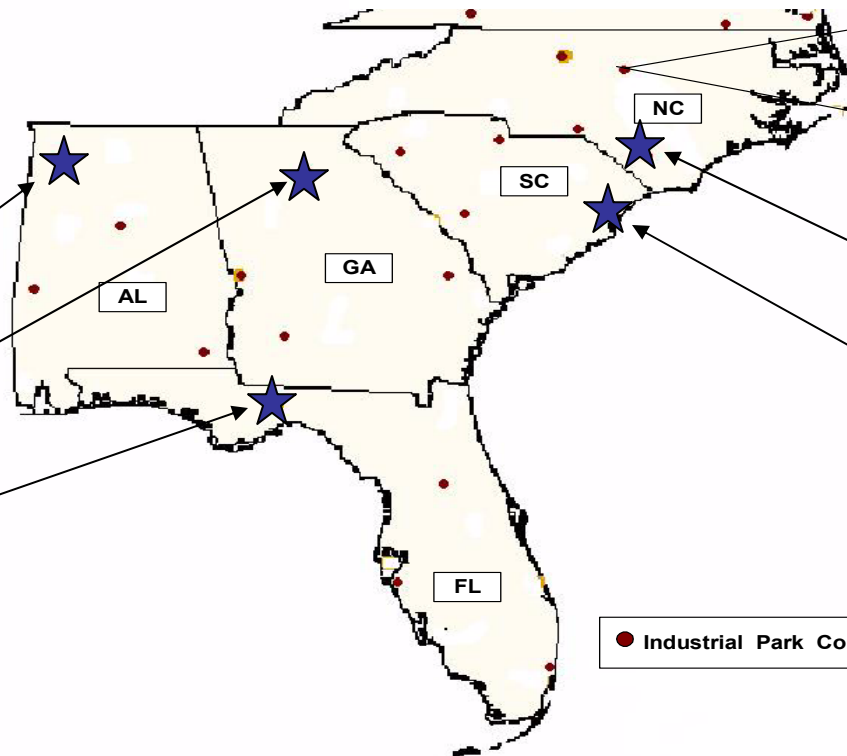
# CONOP Parameters

- As for transitional period human operator will be used to build confidence from our customer and they will monitor at least 5 UAVs and increase as experience gained
- Human in each IP ground operation center
- Shipping Minimum Distance 300 mi
- Shipping Maximum Distance 900 mi



# Selected Industrial Parks Map

## Southeast Region



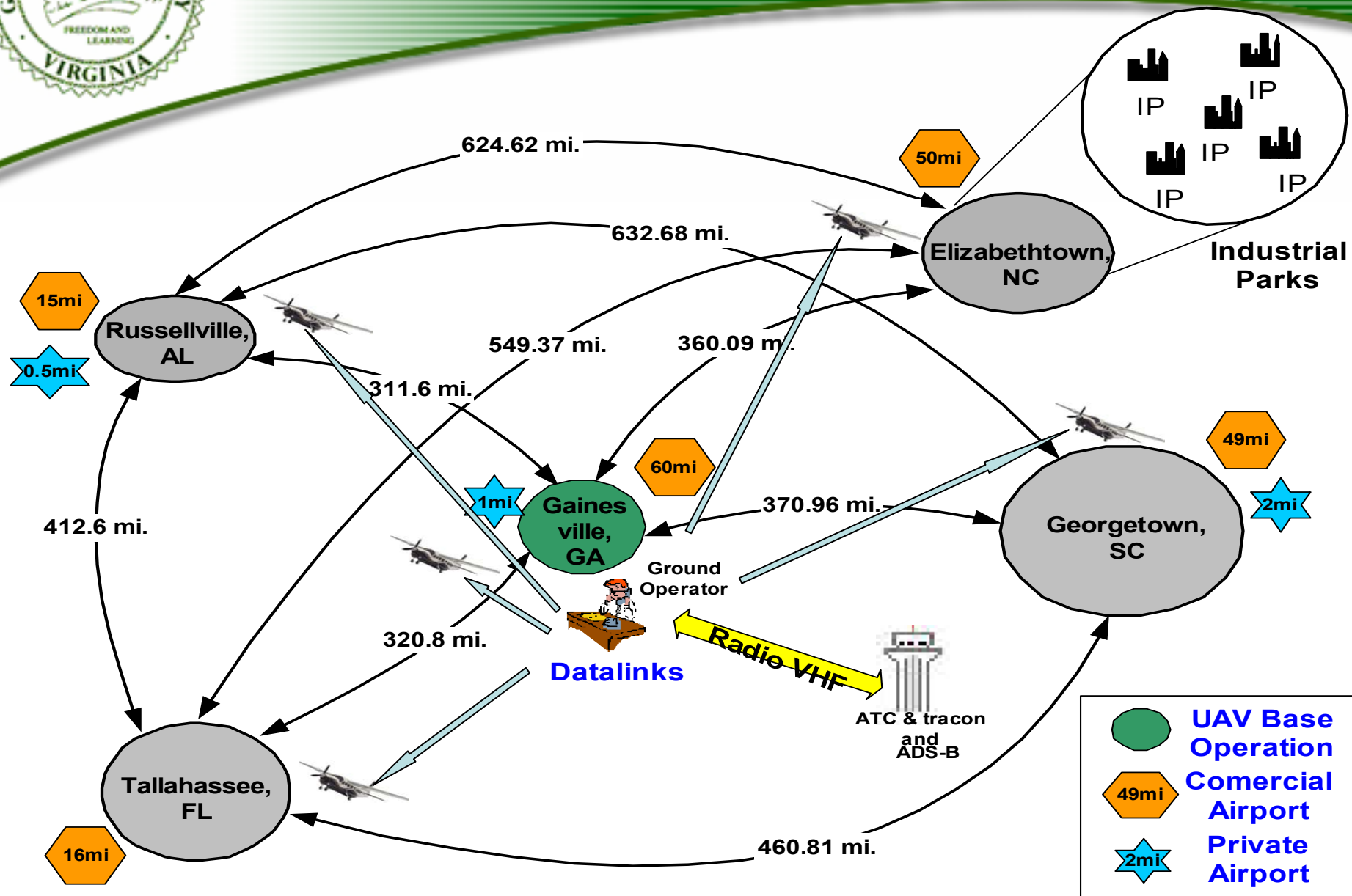
- **Alabama**
  - Russellville
- **Georgia**
  - Gainesville
- **Florida**
  - Tallahassee

- **North Carolina**
  - Elizabethtown
- **South Carolina**
  - Georgetown

● Industrial Park County/Region

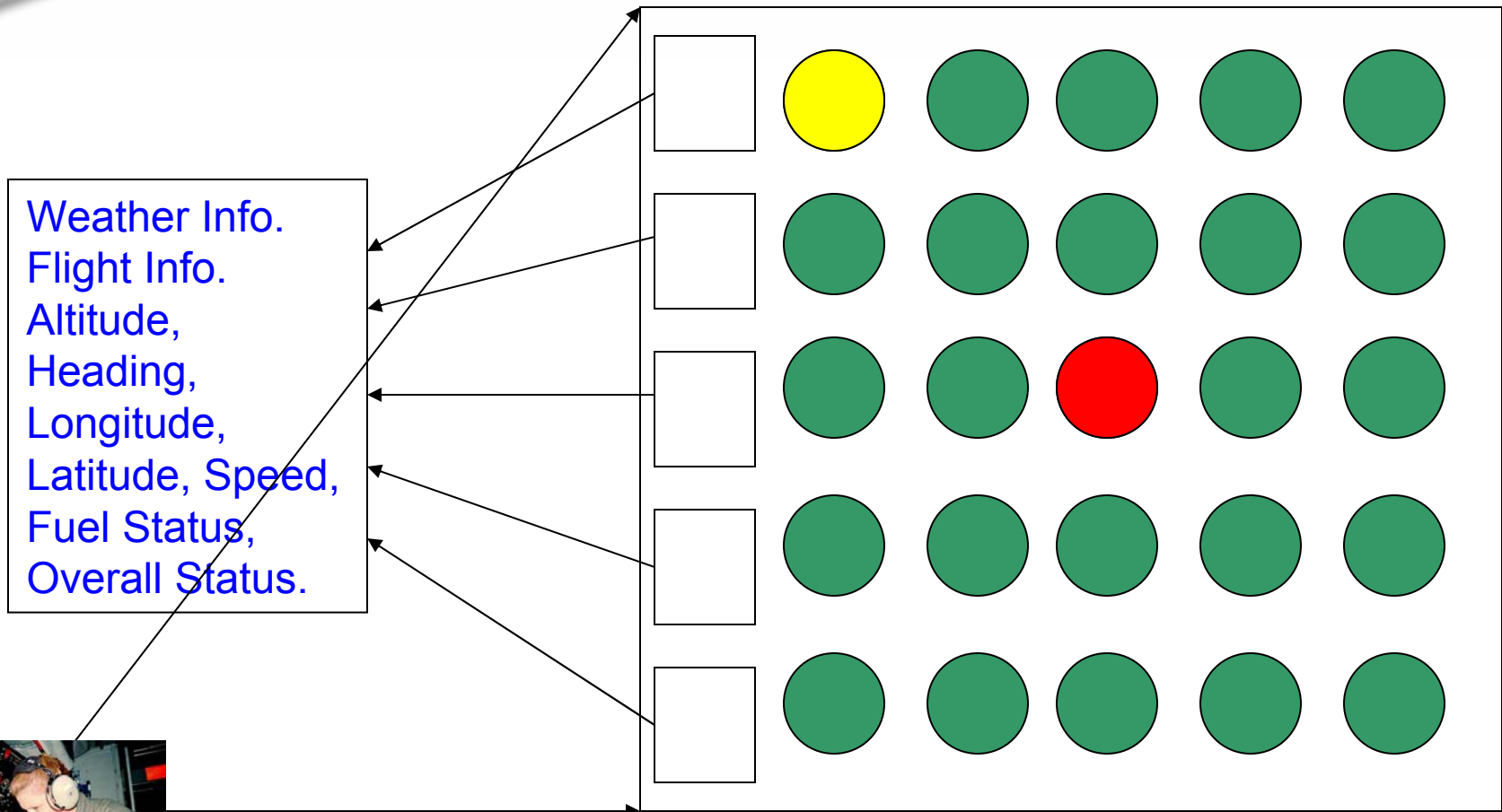


# En Route IP Network Diagram





# Human Operator/Ground Station





# Avionics Architecture



# Architecture

Flight Control  
System  
(GuideStar)

Pilot's Surrogate

INS

Optical System

LORAN

GPS / WAAS

ADS-B

FIS-B

TIS-B

TCAS II

Navigate the UAV  
Aircraft

Provide UAV Fault  
Tolerant Control  
System

Detect Terrain &  
Objects in flight path

En-Route

Take-off

START

Land the UAV Aircraft

Detect Runway

Landing

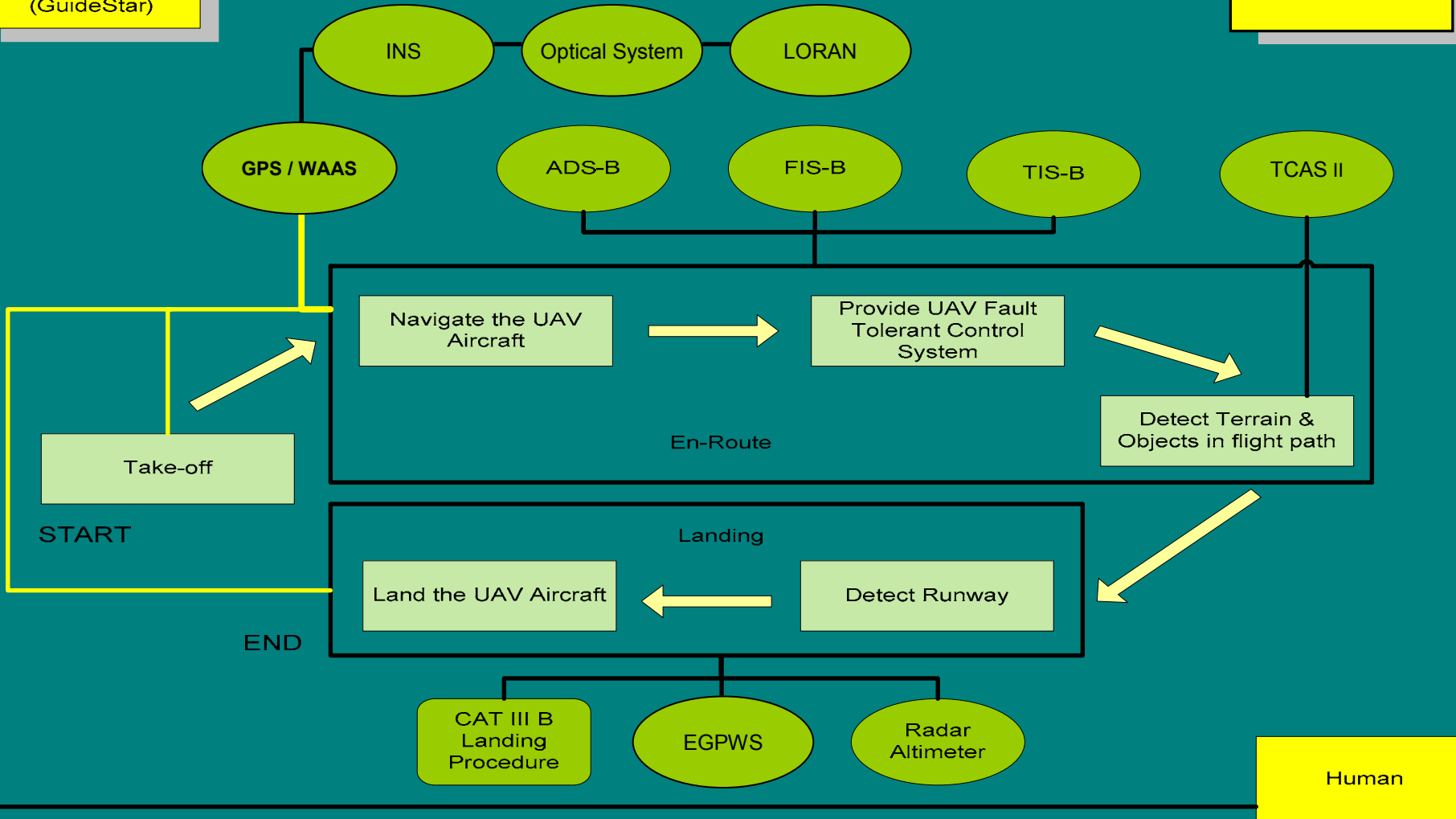
END

CAT III B  
Landing  
Procedure

EGPWS

Radar  
Altimeter

Human





# Data Links

**Mode-C, Mode-S**  
**1030 - 1090 MHz**  
**(UHF/SHF)**  
**FAA, EC**  
**Certified**

**VDL-3**  
**110 MHz (VHF)**  
**FAA Certified**

**VDL-4**  
**Developing**  
**Phase**

**TCAS II**

**ADS-B**

**TIS-B**

**VDL-2**  
**136 MHz**  
**(VHF)**  
**FAA, EC**  
**Certified**

**Remote**  
**Monitoring**

**FIS-B**

**UAT**  
**960 MHz**  
**(VHF)**  
**FAA, EC Certified**

**DME**

**GPS,**  
**WAAS,**  
**LORAN,**  
**INS**

**EGPWS,**  
**VISION,**  
**Radar**  
**Altimeter**

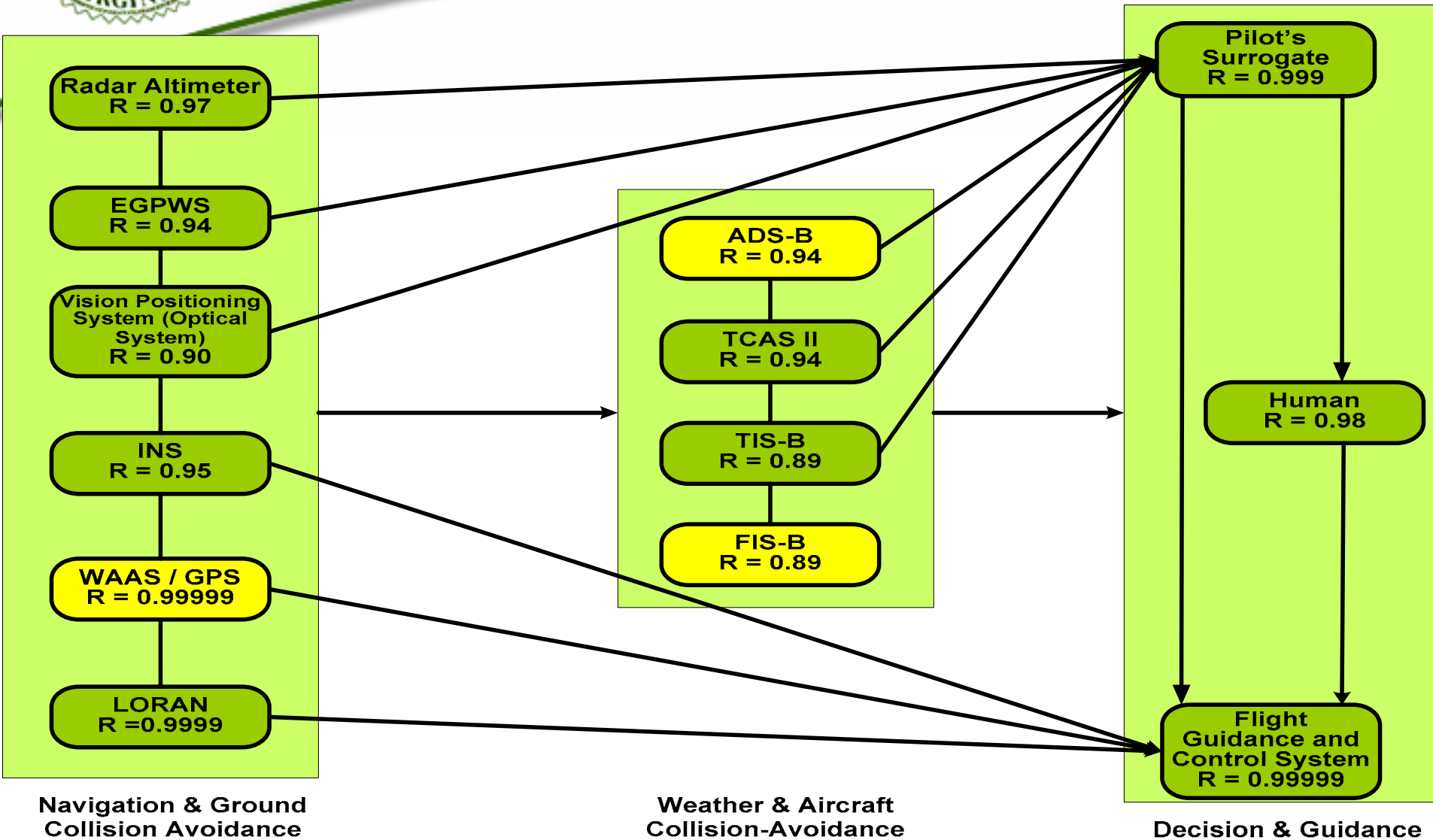




# Reliability Analysis



# Reliability Network





# Diversity and Redundancy

<b>Action</b>	<b>Landing</b>	<b>Collision Avoidance</b>	<b>Surveillance</b>	<b>Data Exchange</b>	<b>Navigation</b>
<b>Data Links &amp; Comps.</b>					
<b>VDL-2 (Remote Monitoring, FIS-B)</b>				X	
<b>VDL-3</b>				X	
<b>VDL-4 (FIS-B, ADS-B, Remote Monitoring)</b>			X	X	
<b>Mode-S (TCAS II, TIS-B, ADS-B)</b>		X	X		
<b>Enhanced Ground Proximity Warning System (EGPWS)</b>	X	X			
<b>Optical System (Vision)</b>	X	X	X		X
<b>LORAN Navigational System</b>	X	X	X		X
<b>Radio-Navigational System (GPS, INS, WAAS)</b>	X	X	X		X



# Calculation of Reliability

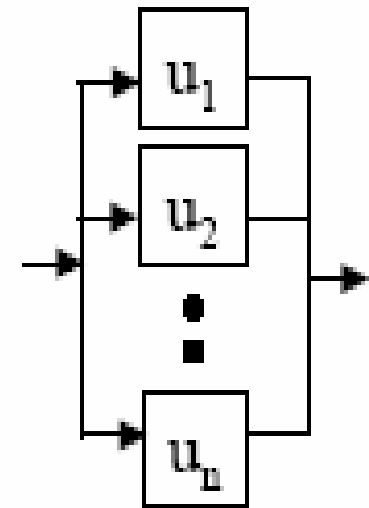
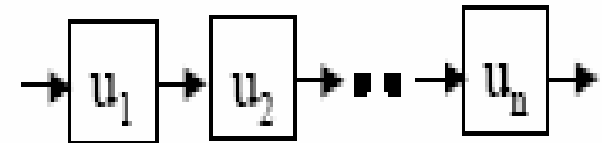
- For  $n$  units connected in series, the system is functioning if all the units are functioning, thus the reliability of the system is

$$R(t) = R_1(t) R_2(t) \dots R_n(t)$$

- Critical system components can never be in series (single string)

- For  $n$  units connected in parallel, the system is functioning if at least one unit is functioning, thus system reliability is

$$R(t) = 1 - [1 - R_1(t)] [1 - R_2(t)] \dots [1 - R_n(t)]$$





# Calculation of Reliability

(cont'd)

- Let  $R(t)$  be the reliability of a system and  $Q(t) = 1 - R(t)$
- The average time between failure is

$$\text{MTBF} = 1/\lambda$$

where  $\lambda$  is the failure rate

- Reliability can be determined for each component using

$$R_n = e^{-(\lambda_n t)}$$

- Ex: System with 3 components

$$R^3 + 3R^2Q + 3RQ^2 + Q^3 = 1$$

where:  $R^3$  represents all three components operating

$Q^3$  represents all three components failing



# Component / Redundancies

## Variable Definitions

	Component / Redundancy										
A	Radar Altimeter				R = Reliability of each component / redundancy						
B	EGPWS				Q = Failure of each component / redundancy						
C	Optical System										
D	INS				$R_{ABCDEF} = 1 - (1 - R_A)(1 - R_B)(1 - R_C)(1 - R_D)(1 - R_E)(1 - R_F)$						
E	WAAS / GPS				$R_{GHIJ} = 1 - (1 - R_G)(1 - R_H)(1 - R_I)(1 - R_J)$						
F	LORAN				$R_Z = R_Z$						
G	ADS-B				$R_{XYZ} = 1 - (1 - R_X)(1 - R_Y)(1 - R_Z)$						
H	TCAS II										
I	TIS-B										
J	FIS-B										
X	Pilot's Surrogate										
Y	Human										
Z	Flight Control System (GuideStar™)										



# System Reliability *with* Human Operator and Pilot's Surrogate

13 components - *with* Human Operator and Pilot's Surrogate

$$R^{13} + 13R^{12}Q + 78R^{11}Q^2 + 286R^{10}Q^3 + 715R^9Q^4 + 1267R^8Q^5 + 1656R^7Q^6 + 1656R^6Q^7 + 1267R^5Q^8 + 715R^4Q^9 + 286R^3Q^{10} + 78R^2Q^{11} + 13RQ^{12} + Q^{13} = 1 = \text{Reliability}$$

Reliability of Entire System

$$R_{ABCDEF} * R_{GHIJ} * R_{XYZ} = 99.99 \%$$



# **Simulation and Analysis Results**





# Results

- System Reliability Analysis
  - Overall System Reliability with Human Operator & Pilot Surrogate = 99.99%
  - At least triple redundancies in each UAVs Operational Criteria such as Navigation, Collision Avoidance and Surveillance which provide the robust, fail soft, and very high reliability
- Landing Simulation
  - None UAVs Superseded by Human Operator
  - 2% Redirected UAVs
  - Ground roll ~ 2400feet



# En-Route Simulation Results

	Number of UAVs per Industrial Park	Time (hours)	Success Delivery	UAV Utilization	Number of Hours Flown per UAV
1 Service Request per city per hour in average	2	720	94%	0.798	575
		2400	95%	0.766	1838
	3	720	99%	0.802	577
		2400	99%	0.764	1834
2 Service Requests per city per hour in average	3	720	89%	0.868	625
		2400	92%	0.846	2030
	5	720	99%	0.894	644
		2400	98%	0.896	2150
3 Service Requests per city per hour in average	5	720	91%	0.914	658
		2400	94%	0.898	2155
	9	720	99%	0.902	649
		2400	99%	0.904	2170



# **Business Case Analysis Results**



# Break Even with 25 Planes

## Breakeven Analysis UAV CARGO SYSTEM

Amounts shown in U.S. dollars

Sales		
Sales price per unit	600.00	
Sales volume per period (units)	83,760	
Total Sales		50,256,000.00
Variable Costs		
Variable costs per unit	388.91	
Total Variable Costs		32,574,713.76
Unit contribution margin	211.09	
Gross Margin		17,681,286.24
Fixed Costs Per Period		
Total Fixed Costs per period		14,724,059.00
Net Profit (Loss)		2,957,227.24

## Results:

**Breakeven Point (units):**

**69,751**

**Sales volume analysis:**

	Year 1	Year 2	Year 3	Year 4	Year 5
Sales volume per period (units)	16,752	33,504	50,256	67,008	83,760
Sales price per unit	600.00	600.00	600.00	600.00	600.00
Fixed costs per period	14,724,059.00	14,724,059.00	14,724,059.00	14,724,059.00	14,724,059.00
Variable costs	6,514,942.75	13,029,885.50	19,544,828.26	26,059,771.01	32,574,713.76
Total costs	21,239,001.75	27,753,944.50	34,268,887.26	40,783,830.01	47,298,772.76
Total sales	10,051,200.00	20,102,400.00	30,153,600.00	40,204,800.00	50,256,000.00
Net profit (loss)	(11,187,801.75)	(7,651,544.50)	(4,115,287.26)	(579,030.01)	2,957,227.24



# Cost Analysis

- Recurring costs are low and Fixed costs are very high, due to:
  - implementation of state-of-the-art technology components and compensating for not having a pilot
- High variable costs make operation expensive
- Return On Investment
  - About 5 years



# Conclusion

- Reliability of the System
- Feasibility of UAV Cargo System Concept
- Business Case
- Usage of Spiral Development
- Possible Complementary Usages



# Acknowledgements

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Director of Aviation Research, Federal Aviation Administration

- **Dr. Lance Sherry**

Research Assistant, George Mason University and Former Director of Business Development, Athena Technologies, Inc.

- **Dr. George Donohue**

Professor of Systems Engineering and Operations Research, George Mason University



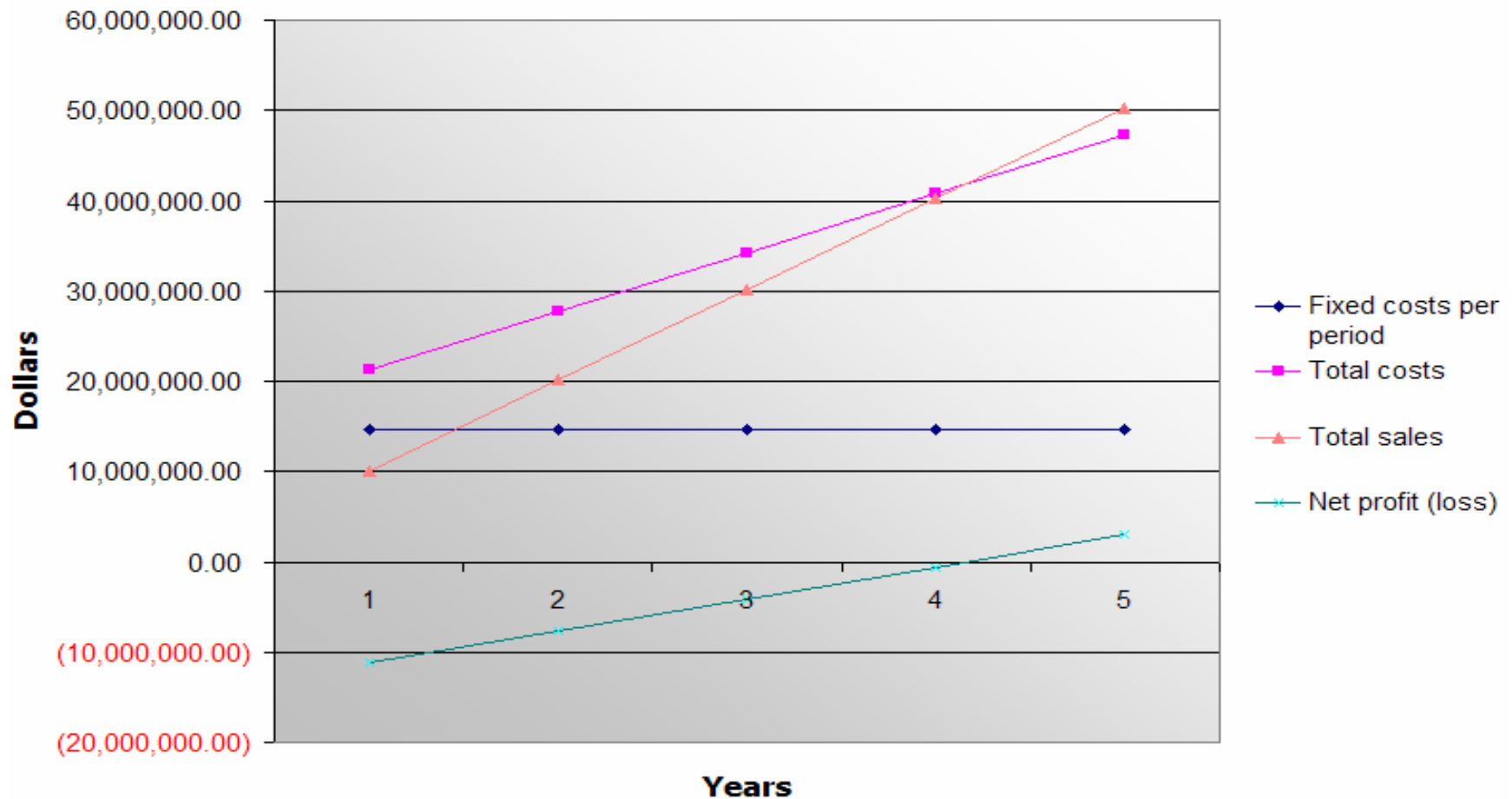
# Questions ?





# Break Even with 25 Planes

**Breakeven Analysis Chart**





# Cost Recommendations

- Buy vs. lease of equipment
- Break even point considering:
  - High equipment costs as a barrier to entry
  - Economies of scale of larger operations
  - Decreasing cost of technology over time
  - Government subsidies made in the public interest
- Financially feasible over time
  - Number of Operator vs. Number of UAVs
  - More Cargo Service Request Characteristics



# Break Even with 45 Planes

## Breakeven Analysis UAV CARGO SYSTEM

Amounts shown in U.S. dollars

Sales		
Sales price per unit	600.00	
Sales volume per period (units)	141,948	
Total Sales		85,168,800.00
Variable Costs		
Variable costs per unit	416.77	
Total Variable Costs		59,159,178.14
Unit contribution margin	183.23	
Gross Margin		26,009,621.86
Fixed Costs Per Period		
Total Fixed Costs per period		22,764,682.20
Net Profit (Loss)		3,244,939.66

## Results:

**Breakeven Point (units):**

**124,239**

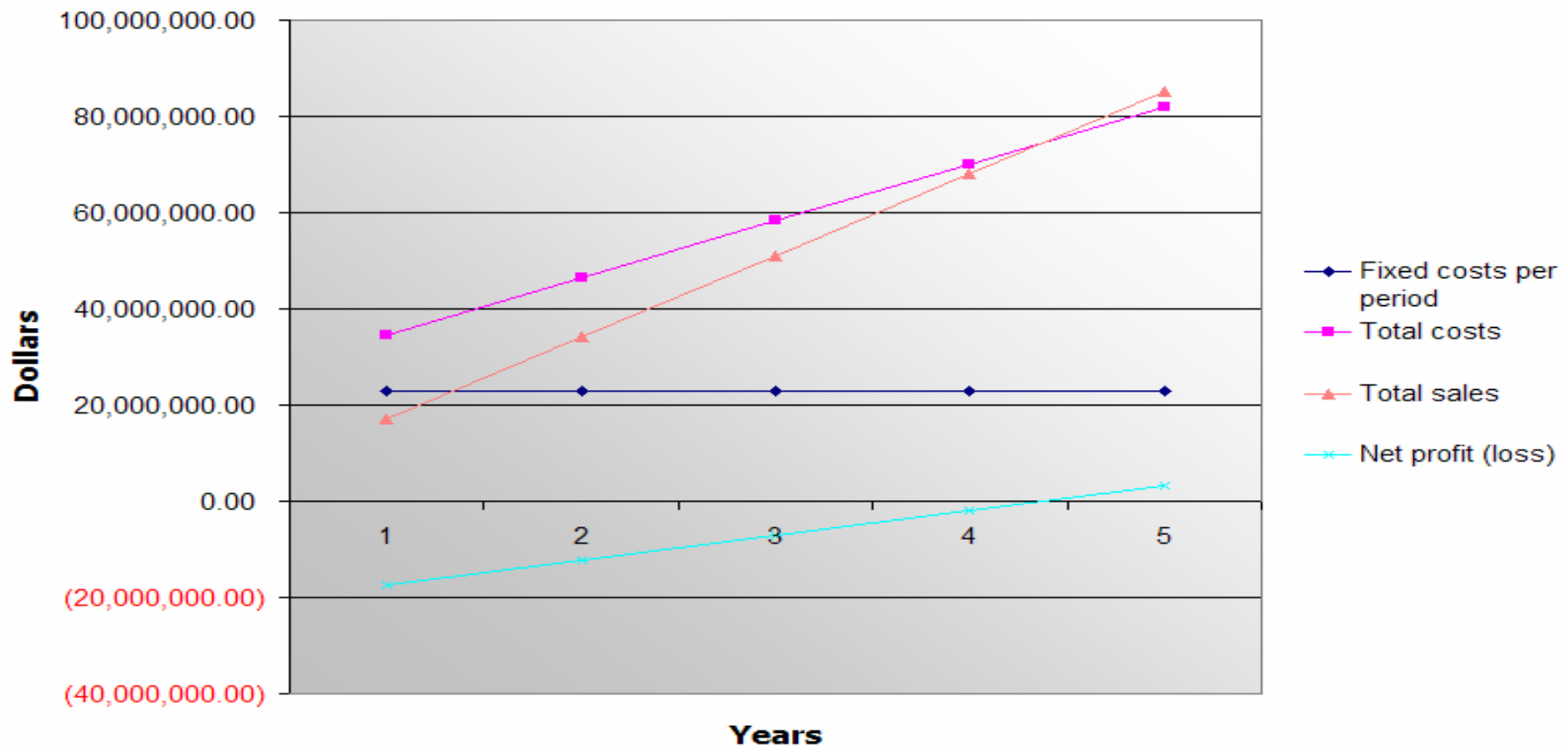
**Sales volume analysis:**

	Year 1	Year 2	Year 3	Year 4	Year 5
Sales volume per period (units)	28,390	56,779	85,169	113,558	141,948
Sales price per unit	600.00	600.00	600.00	600.00	600.00
Fixed costs per period	22,764,682.20	22,764,682.20	22,764,682.20	22,764,682.20	22,764,682.20
Variable costs	11,831,835.63	23,663,671.26	35,495,506.89	47,327,342.52	59,159,178.14
Total costs	34,596,517.83	46,428,353.46	58,260,189.09	70,092,024.72	81,923,860.34
Total sales	17,033,760.00	34,067,520.00	51,101,280.00	68,135,040.00	85,168,800.00
Net profit (loss)	(17,562,757.83)	(12,360,833.46)	(7,158,909.09)	(1,956,984.72)	3,244,939.66



# Break Even with 45 Planes

**Breakeven Analysis Chart**





# Flight Control & Weather

- **Flight Control Guidance System**
  - Interpreting sensor data and making decisions about navigating the airplane
  - Ability to detect and isolate failure part (s), reconfigure aircraft to continue flying without the failed part (s)
- **FIS-B** (Flight Information Service Broadcast)
  - Weather Condition and Provide Icing Altitude



# Concept of Operations

- Size of the Industrial Park (IP)
- Surrogate Plane
- Hypothetical Situation
- Central Operation Center - Human Operator and Interface
- Possible Rare Normal and Abnormal Situation



# Surrogate Plane Specs

- Performance Specification
  - **Certified Ceiling ft/m 25,000/7,620**
  - **Cruise Speed (10,000 ft) knts/km 184/341**
  - **Stall Speed (Ldg) knts/km 61/113**
  - **Take off S.L. ISA Ground Roll ft/m 1,365/416 (50-ft Obs. ft/m 2,420/738)**
  - **Landing S.L. Ground Roll ft/m 950/290 (50-ft. Obs. ft/m 1,795/547)**
  - **Maximum Useful Load**
    - 3,500 lbs for 100 miles trip
    - 1,500 lbs for 900 miles trip





# Navigation

- **WAAS** (Wide Area Augmentation System) / GPS
  - Provides correction and integrity signals for standard GPS signals
- **LORAN** (LOnge-range Radio Aid to Navigation system)
  - A long-range radio navigation position-fixing system consisting of an array of fixed stations that transmit precisely synchronized signals to mobile receivers.
- **INS** (Inertial Navigational System)
  - Backup for the GPS
- **Vision Position System** (Optical System)
  - Calculates the Altitude to Runway and Surveillance of the Surroundings





# Break Even with 15 Planes

## Breakeven Analysis

## 15 Planes

### UAV CARGO SYSTEM

Amounts shown in U.S. dollars

#### Sales

Sales price per unit	650.00	
Sales volume per period (units)	42,300	
<b>Total Sales</b>		<b>27,495,000.00</b>

#### Variable Costs

Variable costs per unit	414.50	
<b>Total Variable Costs</b>		<b>17,533,503.65</b>
<b>Unit contribution margin</b>	<b>235.50</b>	
<b>Gross Margin</b>		<b>9,961,496.35</b>

#### Fixed Costs Per Period

<b>Total Fixed Costs per period</b>	<b>9,758,627.40</b>
<b>Net Profit (Loss)</b>	<b>202,868.95</b>

## Results:

**Breakeven Point (units):**

**41,439**

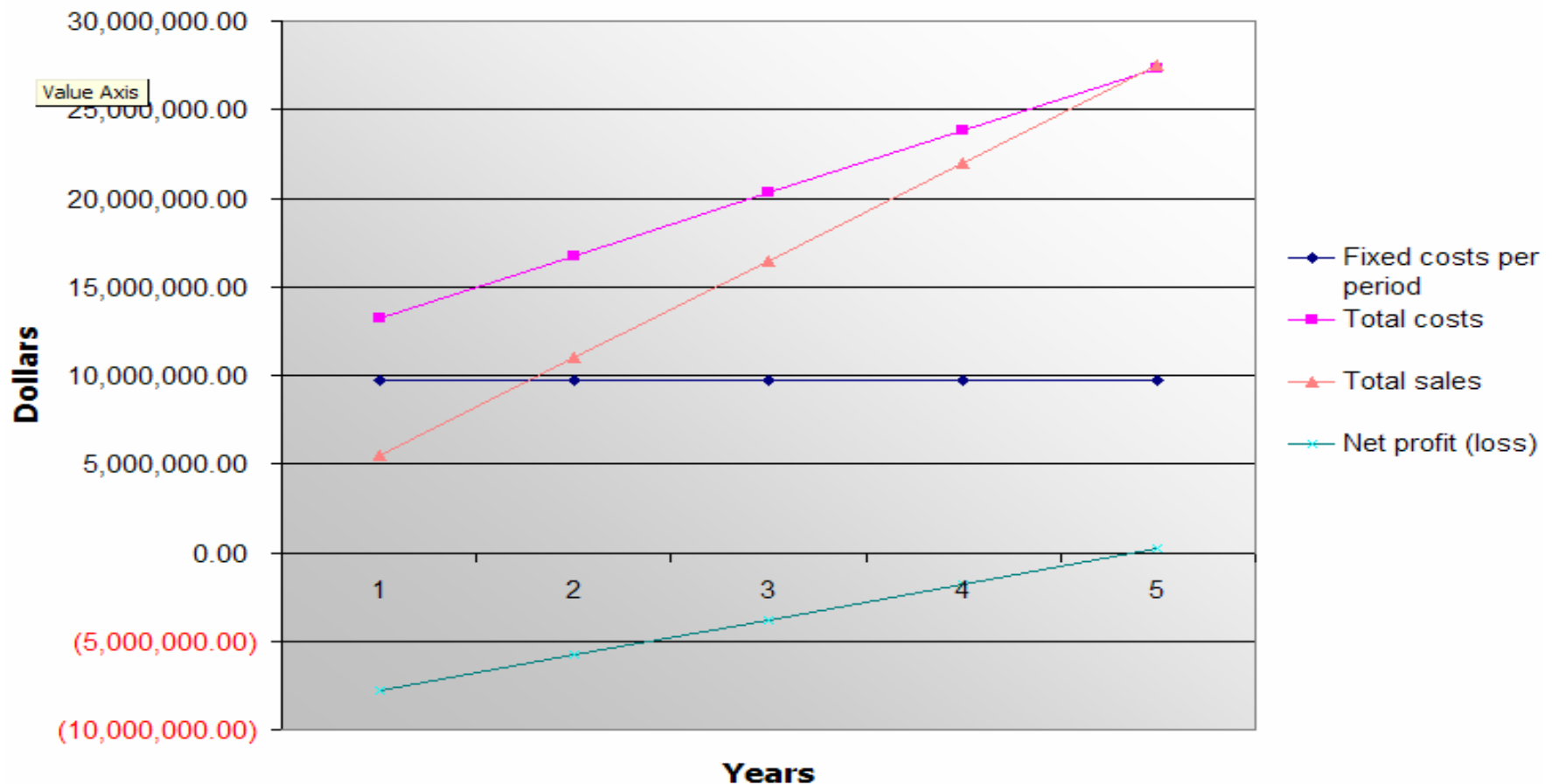
**Sales volume analysis:**

	Year 1	Year 2	Year 3	Year 4	Year 5
Sales volume per period (units)	8,460	16,920	25,380	33,840	42,300
Sales price per unit	650.00	650.00	650.00	650.00	650.00
Fixed costs per period	9,758,627.40	9,758,627.40	9,758,627.40	9,758,627.40	9,758,627.40
Variable costs	3,506,700.73	7,013,401.46	10,520,102.19	14,026,802.92	17,533,503.65
Total costs	13,265,328.13	16,772,028.86	20,278,729.59	23,785,430.32	27,292,131.05
Total sales	5,499,000.00	10,998,000.00	16,497,000.00	21,996,000.00	27,495,000.00
Net profit (loss)	(7,766,328.13)	(5,774,028.86)	(3,781,729.59)	(1,789,430.32)	202,868.95



# Break Even with 15 Planes

**Breakeven Analysis Chart**





# Collision Avoidance & Surveillance

- **ADS-B** (Automatic Dependant Surveillance Broadcast)
  - Navigation State Vector Surveillance
- **TIS-B** (Traffic Information Service Broadcast)
  - Air and Ground Collision Avoidance
- **TCAS II** (Traffic Collision Avoidance System)
  - Last Resort Air Collision Avoidance System
- **EGPWS** (Enhanced Ground Proximity Warning System)
  - Terrain awareness and alerting system
- **Radar Altimeter**
  - Terrain Collision Avoidance & Measures the height of the main wheels above touchdown



# Additional Data Links

- **VDL:** Digital data link that operate in the VHF frequency band:
  - **Mode 2:** Data, 31.5 Kb, Carrier Sense Multiple Access (CSMA) technology
  - **Mode 3:** Voice and data, 31.5 Kb, Time Division Multiple Access (TDMA) technology
  - **Mode 4:** Data, 19.5 Kb, TDMA technology with self-organizing function (STDMA)
- **Mode-S:** Data, CSMA, 2 Mbits, 1090/1030 MHz (UHF/SHF)



# Landing Sequence

Plane will descend in a glide slope of 3 degree with WAAS always active.

**WAAS** – Center the Runway with Plane.

**INS** and **EGPWS** activate to find the runway distance and center it for successful landing.

**Vision** and **Radar Altimeter** activate to calculate the vertical distance from runway.

**Use of Flight Control Unit (GuideStar)** the plane is able to perform a flare maneuver.

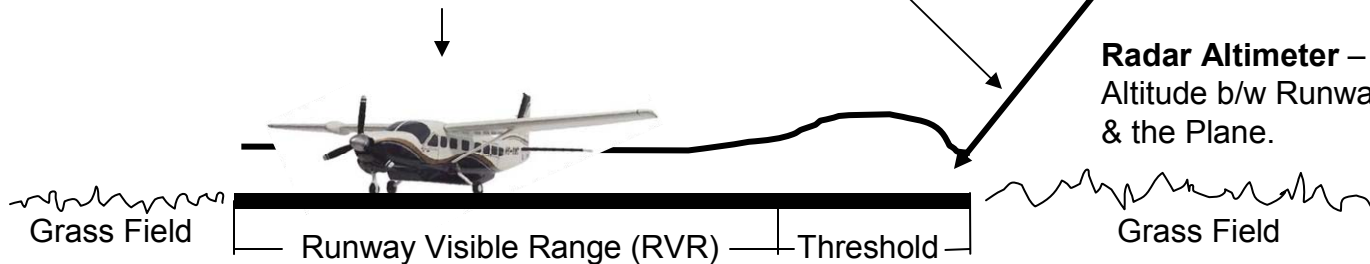
**Enhanced Ground Proximity Warning System (EGPWS)** – Situation of the Environments.

**Optical System (Vision)** – Eye for the Plane.

**Flight Control Unit (GuideStar)** – Brain for the Plane.

**Radar Altimeter** – Altitude b/w Runway & the Plane.

**Ground Roll (Stopping Distance)** is calculated.

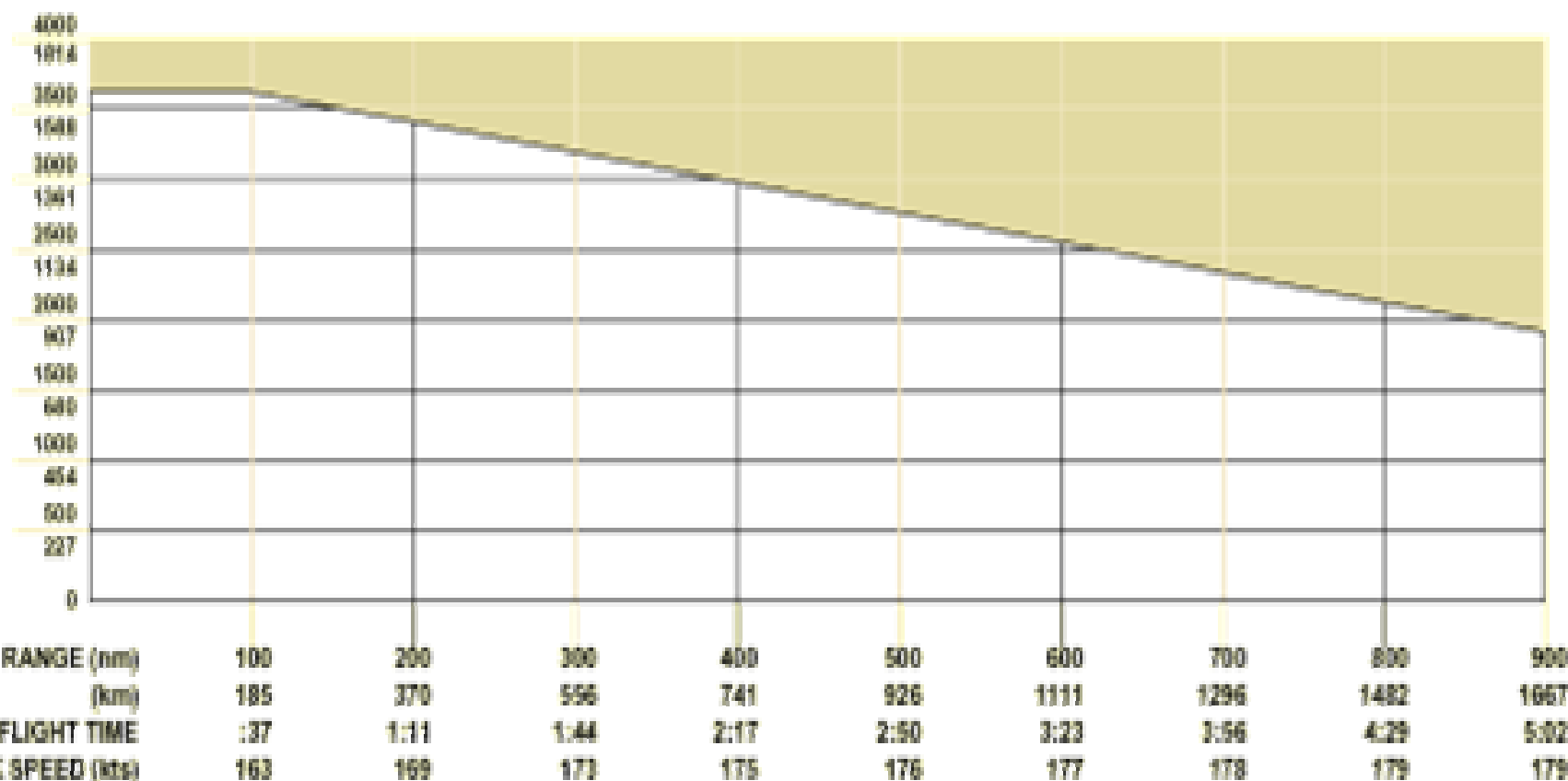


**Flare Height of 30 ft. with 75 knots of Speed to do a Successful Flare Maneuver.**



# CONOP Parameters

**PAYLOAD** Caravan (without pod)  
 (lbs) BOW = 4650 lb/2109 kg  
 (kg) (Includes 170-lb pilot & 45-minute fuel reserve)





# Rare Normal Scenario

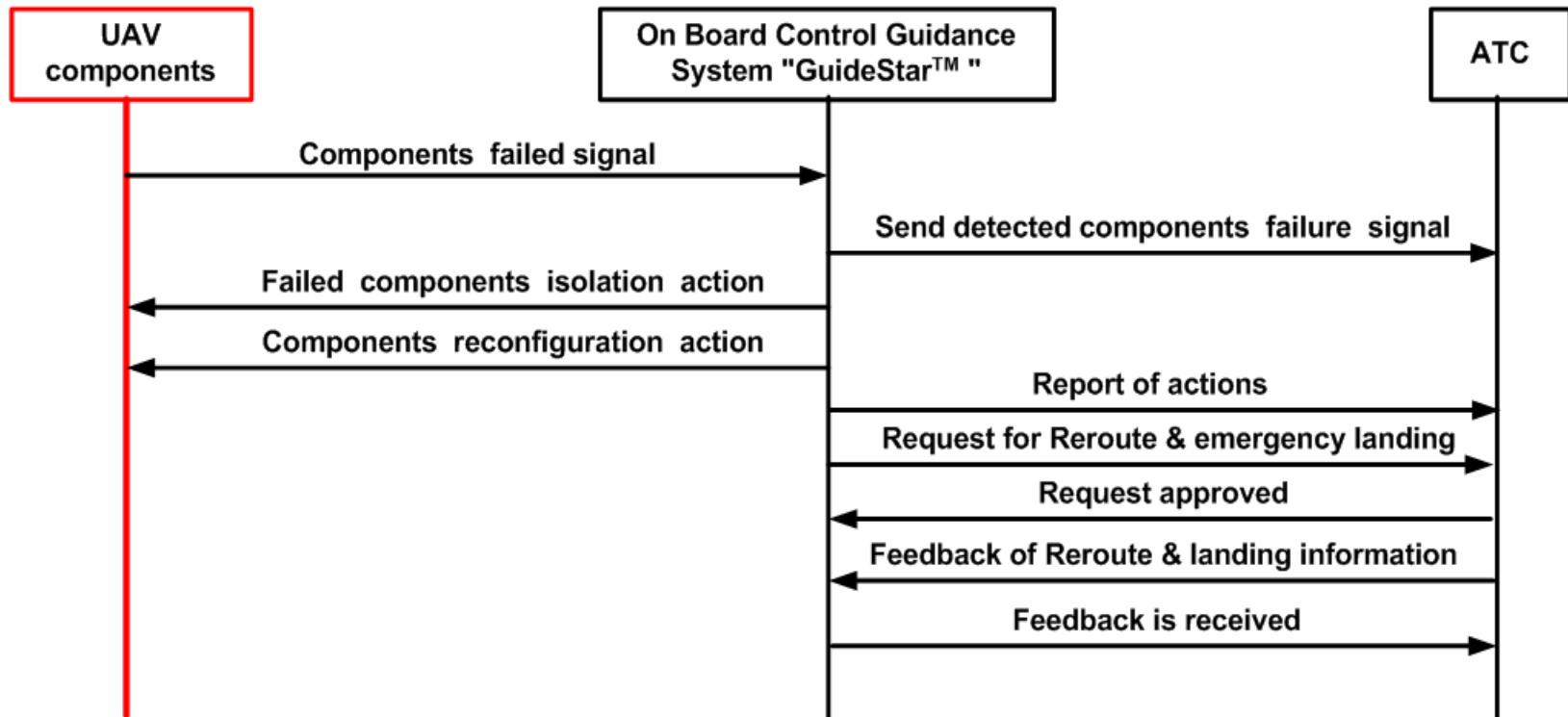


Figure 7. Rare Normal Scenario – when some of UAV components have failed





# Utility Function

- **UAV Cargo System Utility Function Questionnaire**
- How much would you pay to reduce the cost of the system? Example: Would pay \$10M to improve the cost from \$20M to \$8M.

<u>Cost</u>	<u>Value Score</u>	Dollar Amount: \$6M < \$ Optimal < \$10M
\$4 M	100	
\$6 M	70	
\$8 M	30	
\$10 M	0	

- How much would you pay to improve the operational performance of the system?

<u>Operational Performance</u>	<u>Value Score</u>	Optimal Hours Amount: 8 Hrs/day to deliver
8 hrs/day	100	
12 hrs/day	50	
24 hrs/day	0	

- How much would you pay to increase the safety of the system?

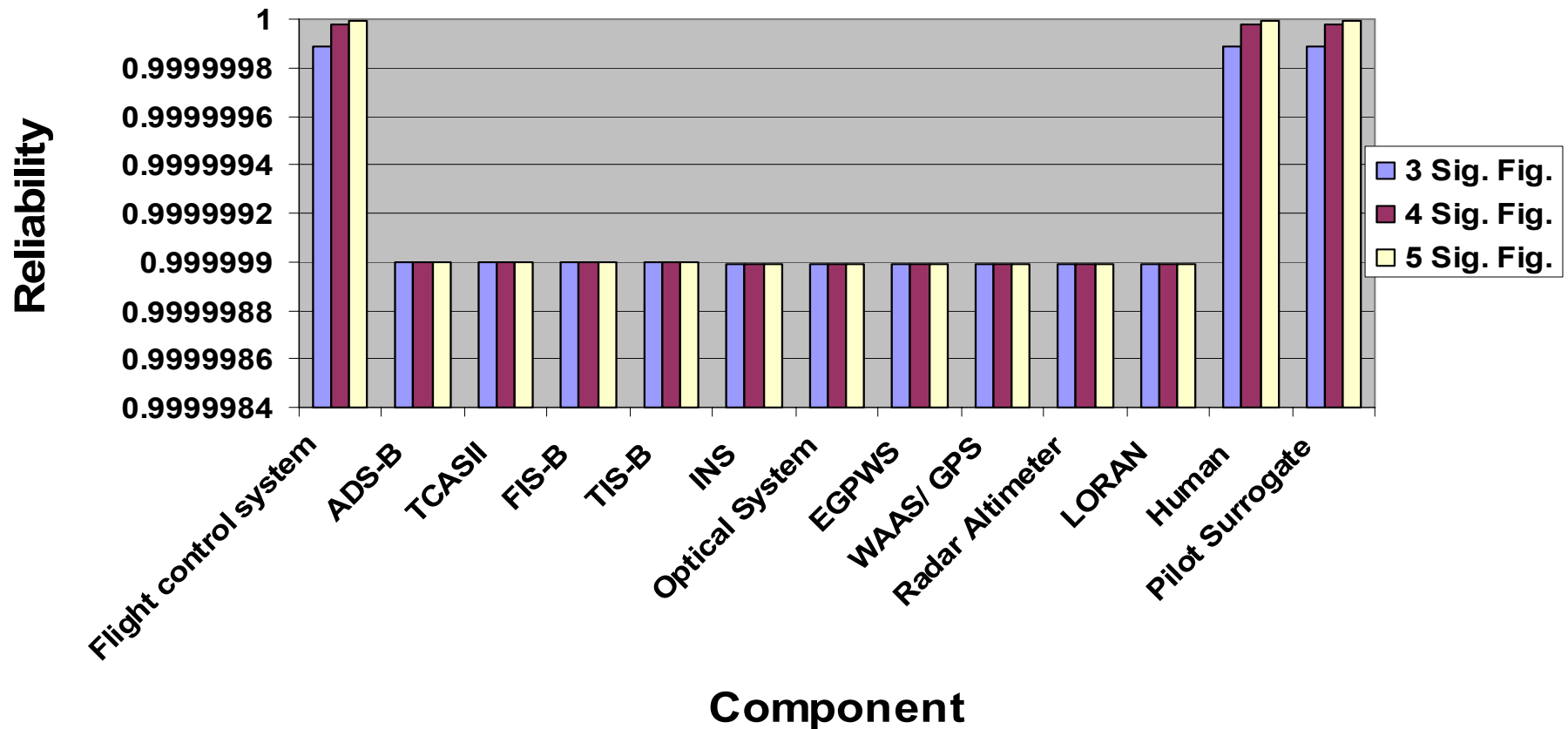
<u>Safety</u>	<u>Value Score</u>	Optimal Reliability Amount: 99.999% Reliability
99.999% reliable	100	
99.99% reliable	95	
99.9% reliable	50	
98% reliable	0	





# Sensitivity Analysis

## Sensitivity Analysis: Comparison of Significant Figures with Human Factors





# Pilot's Surrogate

Utilizes sensor fusion and guidance technologies enabling the optimal blending of redundant sensors during the various phases of a mission, and the generation of appropriate guidance commands for steering the vehicle.

## Collision Avoidance Sensors

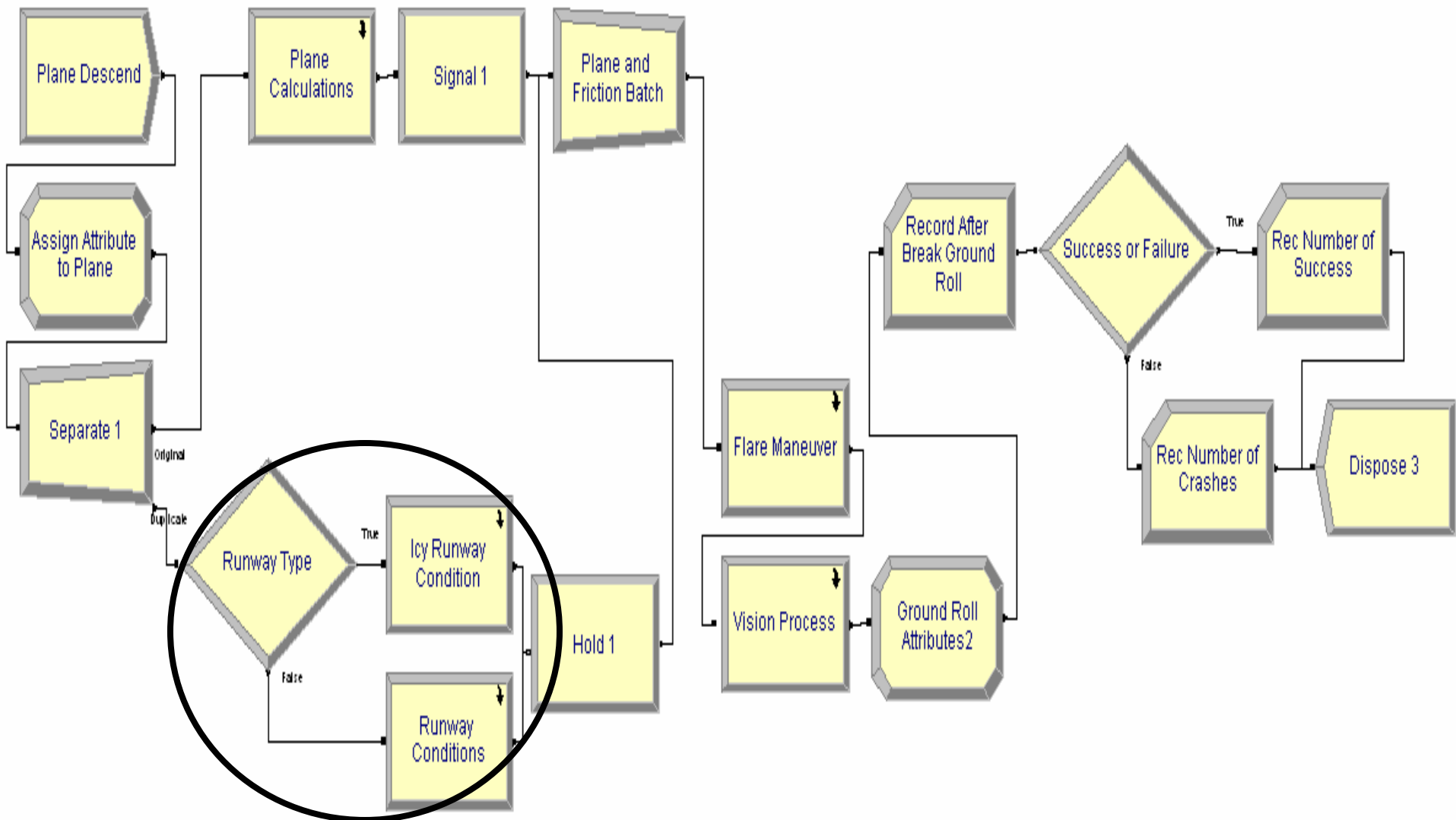
Optical System  
EGPWS  
Radar Altimeter

ADS-B  
TIS-B  
TCAS II

Voting Algorithm

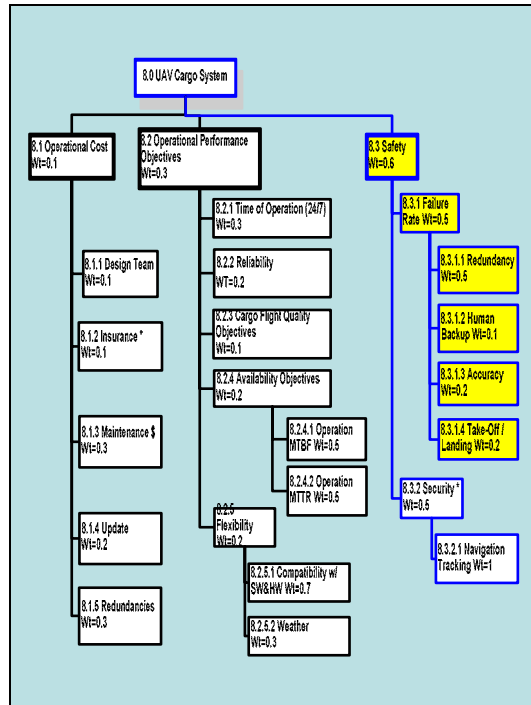


# ARENA Simulation





# Detailed Value Hierarchy



## 8.0 UAV Cargo System

### 8.3 Max. Safety Wt=0.6

#### 8.3.1 Failure Rate Wt=0.5

##### 8.3.1.1 Redundancy Wt=0.5

##### 8.3.1.2 Human Backup Wt=0.1

##### 8.3.1.3 Accuracy Wt=0.2

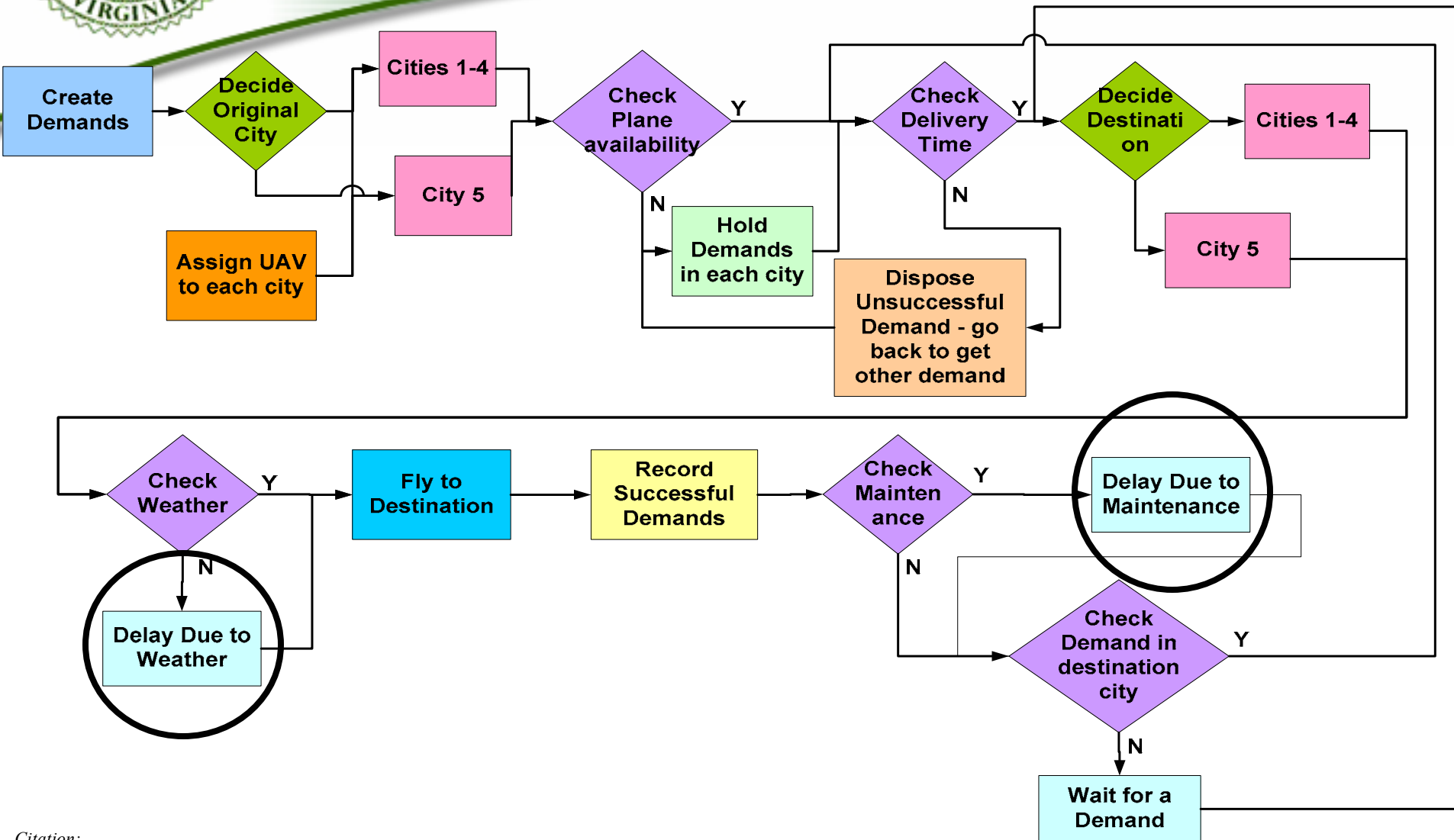
##### 8.3.1.4 Take-Off / Landing Wt=0.2

#### 8.3.2 Security \* Wt=0.5

##### 8.3.2.1 Navigation Tracking Wt=1



# ARENA Simulation Logic





# Surrogate Plane Specs

- Performance Specification

- **Certified Ceiling ft/m 25,000/7,620**

- **Cruise Speed (10,000 ft) knts/km 184/341**

- **Stall Speed (Ldg) knts/km 61/113**

- **Takeoff S.L. ISA Ground Roll ft/m 1,365/416 (50-ft Obs. ft/m 2,420/738)**

- **Landing S.L. Ground Roll ft/m 950/290 (50-ft. Obs. ft/m 1,795/547)**

- **Maximum Useful Load ~ 3,500 lbs for 100 miles trip**

- **Maximum Weights lbs/kg Ramp 8,785/3,985**

- **Standard Empty Weight lbs/kg 4,285/1,944**

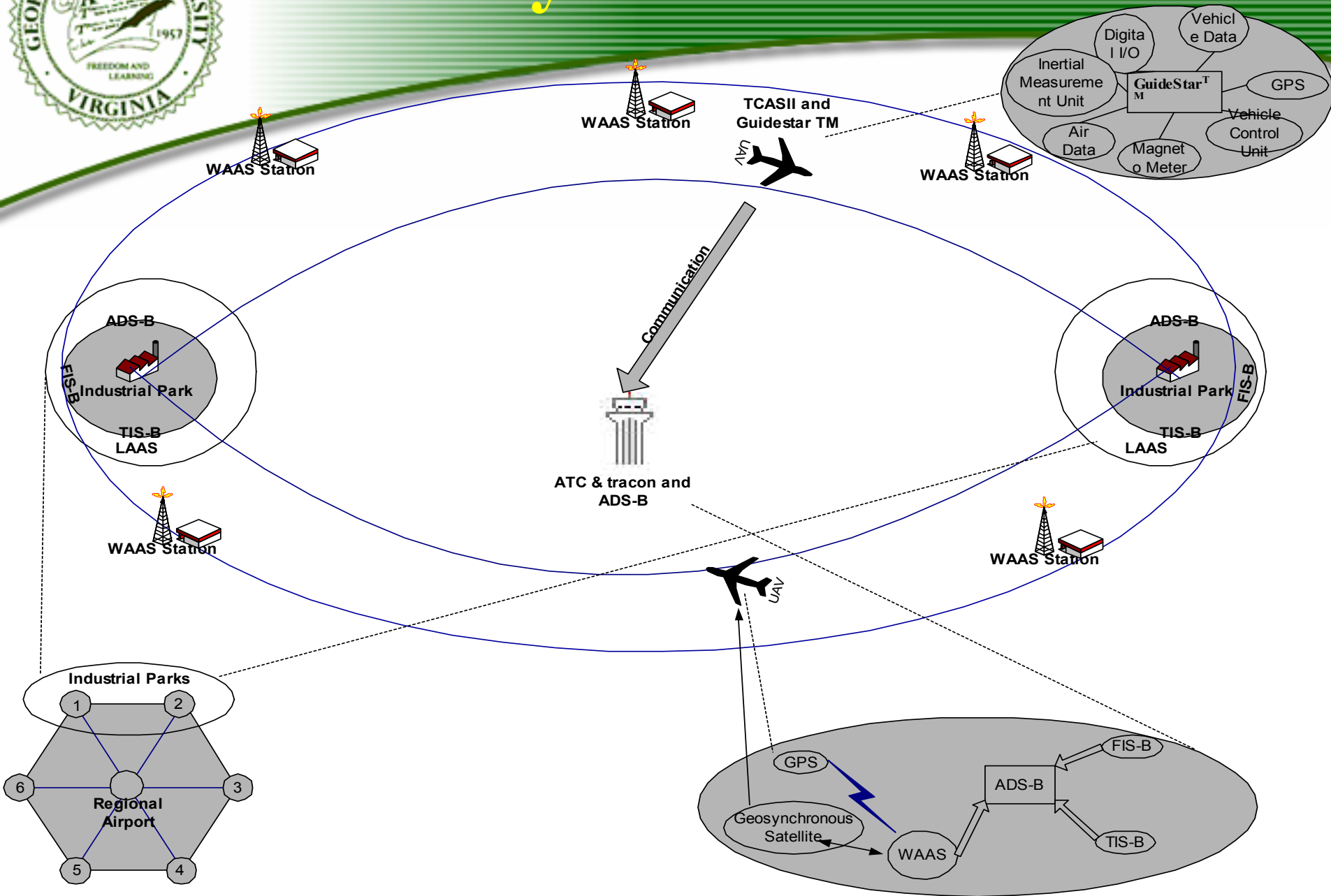


# Parameters

- Individual reliability, MTBF, and availability of components are set by manufacturing specifications
- A collection of *n components* related to the UAV system



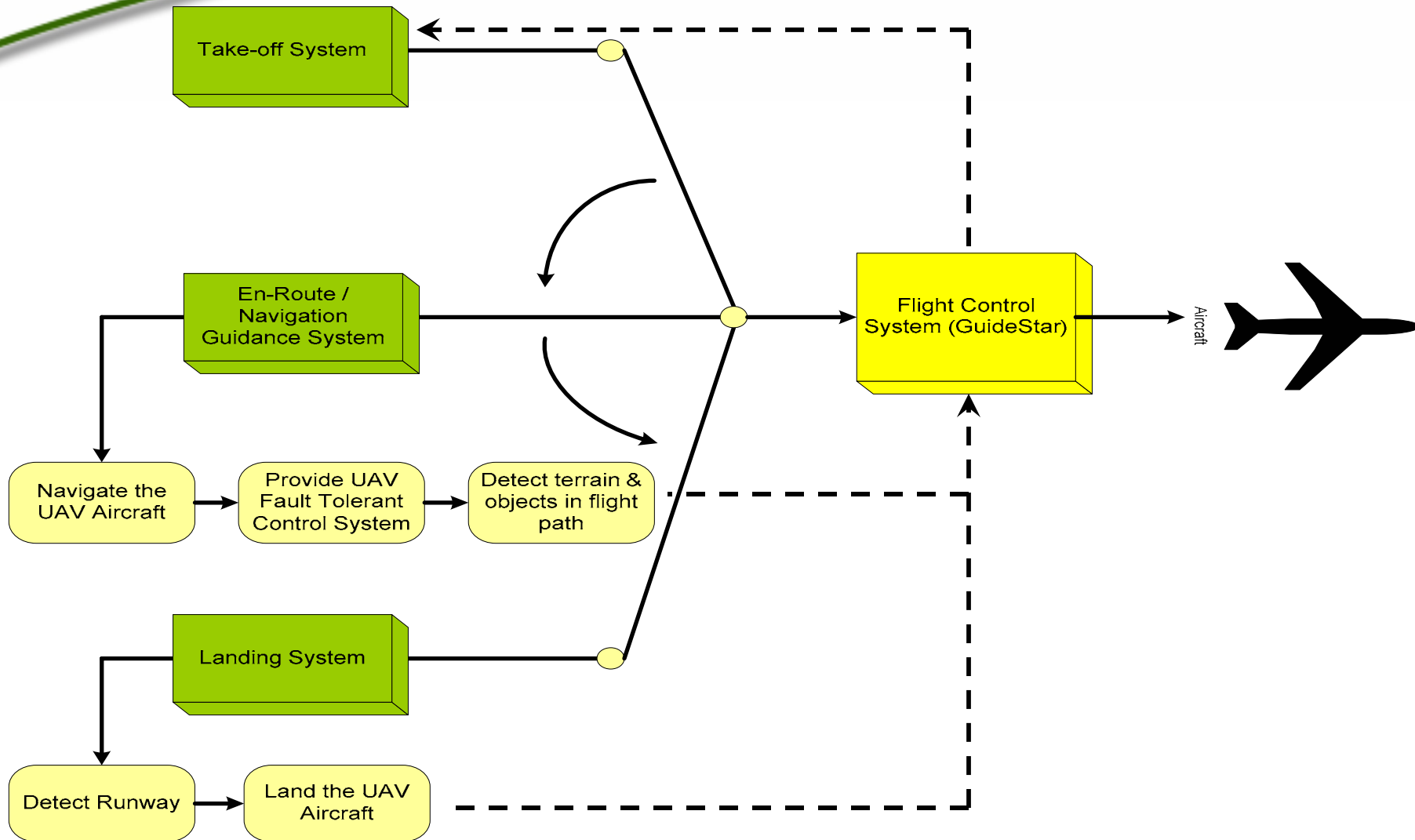
# System Overview







# Flow Diagram – Flight Process

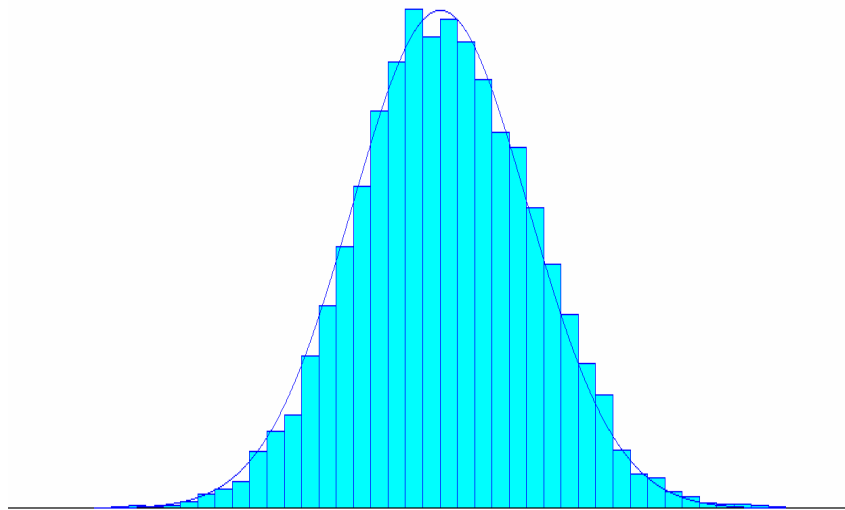




# Statistical Distributions

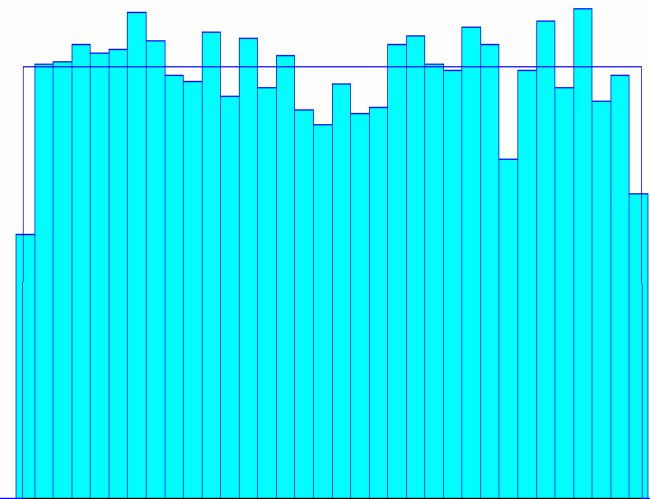
## Runway Condition

– Icy Condition Friction:



$\text{NORM}(0.0826, 1.27)$

– Dry and Wet Condition Friction:



$\text{UNIF}(.5, .8)$

Wet

Dry

Citation:

Trani, Antonio. SATSLab, Virginia Tech. Pg 232. "Transportation System Baseline Assessment Study". Revision 1.0, May 30, 2002.



# Statistical Distributions

- Time of arrival =
  - Triangular Distribution (15, 30, 40)
- Entities / arrival =
  - Uniform Distribution (1, 5)
- Delay time for Loading & Unloading =
  - Gamma Distribution ( beta = 7.7, alpha = 1.6)
- Delay time for Maintenance (100 Hrs) =
  - Weibull Distribution ( beta = 12.2, Std. Dev = 1.98)
- Delay time for Weather =
  - Normal Distribution ( mean = 46, Std. Dev = 9)



# Estimated Simulation Hours

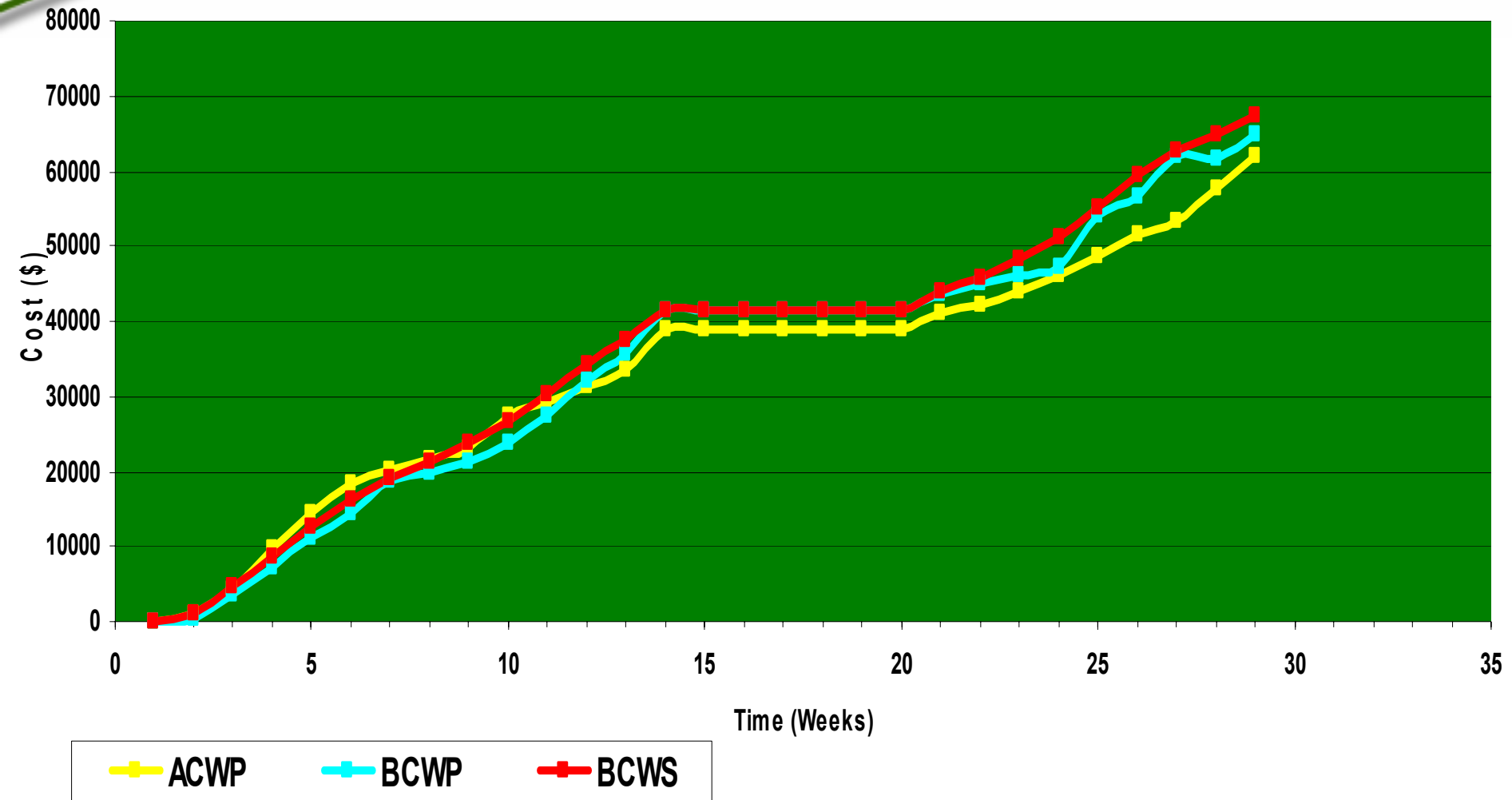
- To Date:
  - 150 Hours x \$20.00 = \$3,000.00
  - 90 % Overall Simulation Completed
- Overall:
  - 180 Hours x \$20.00 = \$3,600.00
  - To complete the simulation plan, estimated number of hours left = 30 Hours



# EVM



## Earned Value Management (EVM) for SYST 490





# Tornado Chart



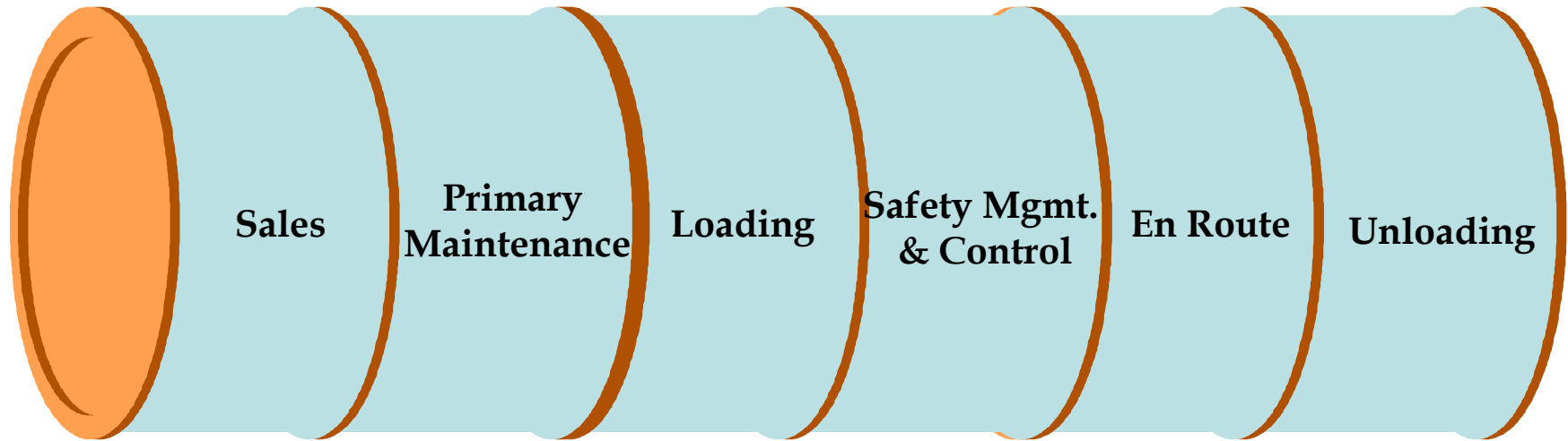
# Key Cost Factors

- Start up costs including:
  - Basic operating infrastructure (OH)
  - Capital expenditures (airplanes, systems and other major equipment)
- Ongoing fixed and variable operating costs





# Operational Costs of Primary Processes





# Key Revenue Drivers

- Understanding our market
  - High end corporate users willing to pay a premium for a few hour's difference
    - Is the business case justifiable solely based on potential demand?
  - Government willingness to support a system that reduces congestion and increases safety
- Defining pricing policy
  - Weight vs. size



# Points to Consider

- Buy vs. lease of equipment
- Break even point considering:
  - High equipment costs as a barrier to entry
  - Economies of scale of larger operations
  - Decreasing cost of technology over time
  - Govt. subsidies made in the public interest



# Cost Template

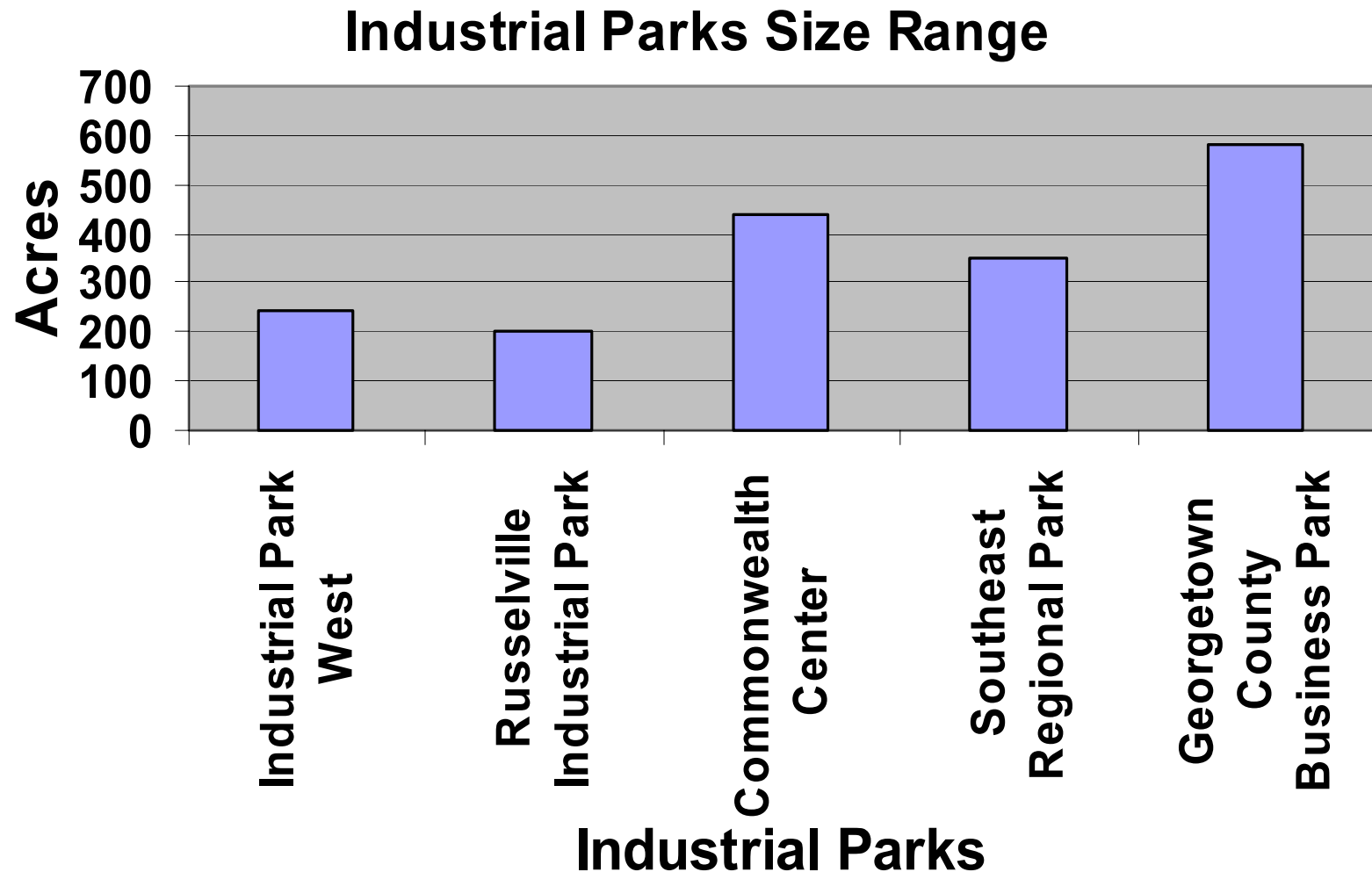
Primary Process	Associated Costs	Annual Fixed Cost	Annual Variable Cost	Annual Total Cost	Start up costs
<b>Flight</b>					
	ADS-B			\$ -	\$ 150,000
	EGPWS			\$ -	\$ 137,660
	Guidestar			\$ -	\$ 325,000
	LORAN			\$ -	\$ 3,945
	ModeS Transponder			\$ -	\$ 24,175
	Radar Altimeter			\$ -	\$ 31,950
	TCASII			\$ -	\$ 140,000
<b>Infrastructure</b>					
	Airplanes	\$ 1,097,500		\$ 1,097,500	\$ -
<b>Loading</b>					
	Loaders			\$ -	
<b>Primary Maintenance</b>					
	Flight mechanics	\$ 898,560		\$ 898,560	
	Flight system analysis			\$ -	
	Refuel			\$ -	
<b>Sales</b>					
	Advanced online order	\$ -		\$ -	
	Basic communications	\$ 24,000		\$ 24,000	
	Ground operator to determine	\$ 811,637		\$ 811,637	
	Ground operator to input the	\$ -		\$ -	
	Order entry Personnel	\$ 149,760		\$ 149,760	
	Order entry Personnel sends	\$ -		\$ -	
	Personnel to check database for	\$ -		\$ -	
	Plane officially scheduled	\$ -		\$ -	
<b>Unloading</b>					
	Loaders	\$ 318,240		\$ 318,240	\$ -
<b>Grand Total</b>		\$ 3,299,697		\$ 3,299,697	\$ 812,730



# Cost Results

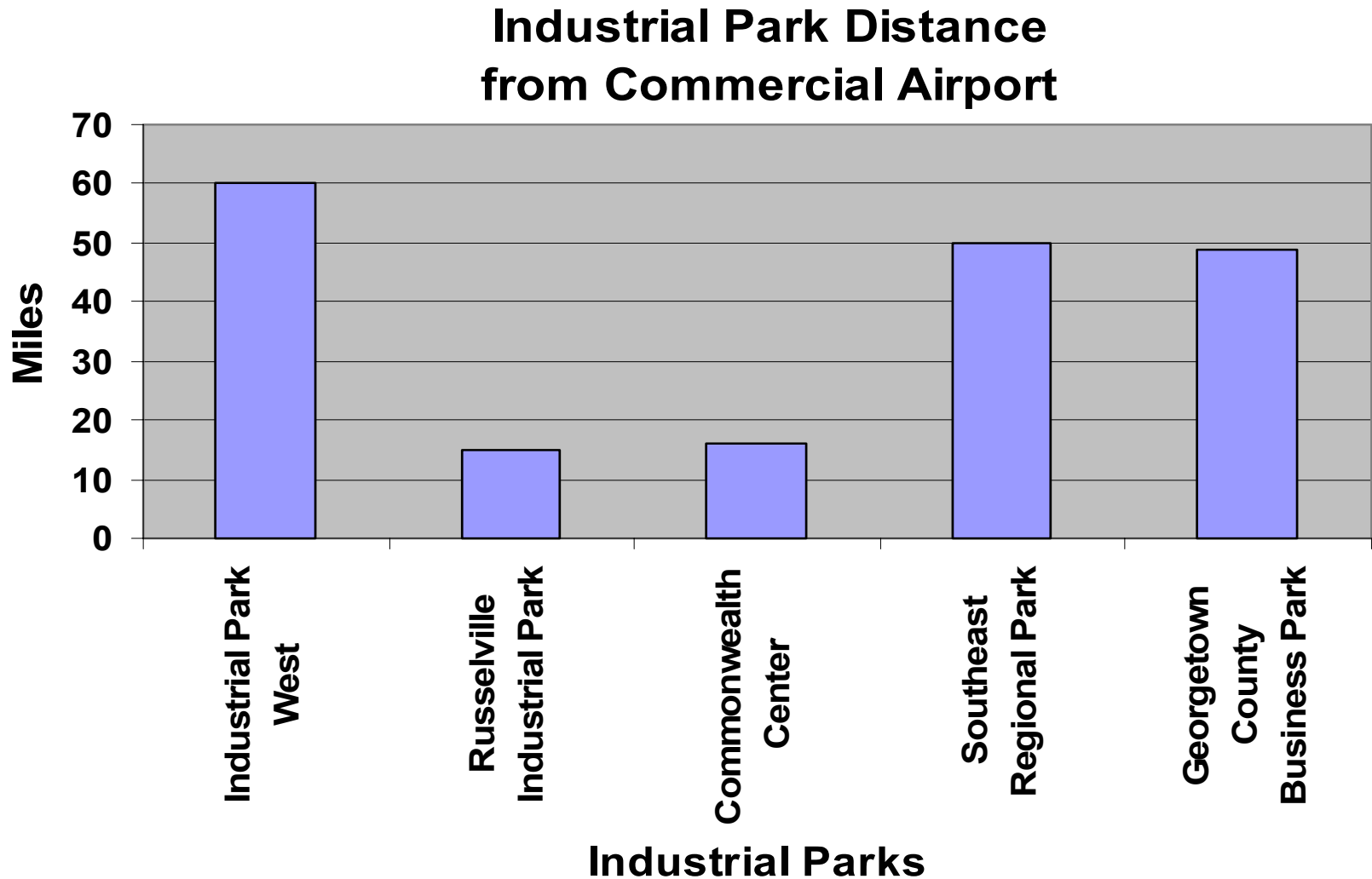


# Industrial Park Size Range





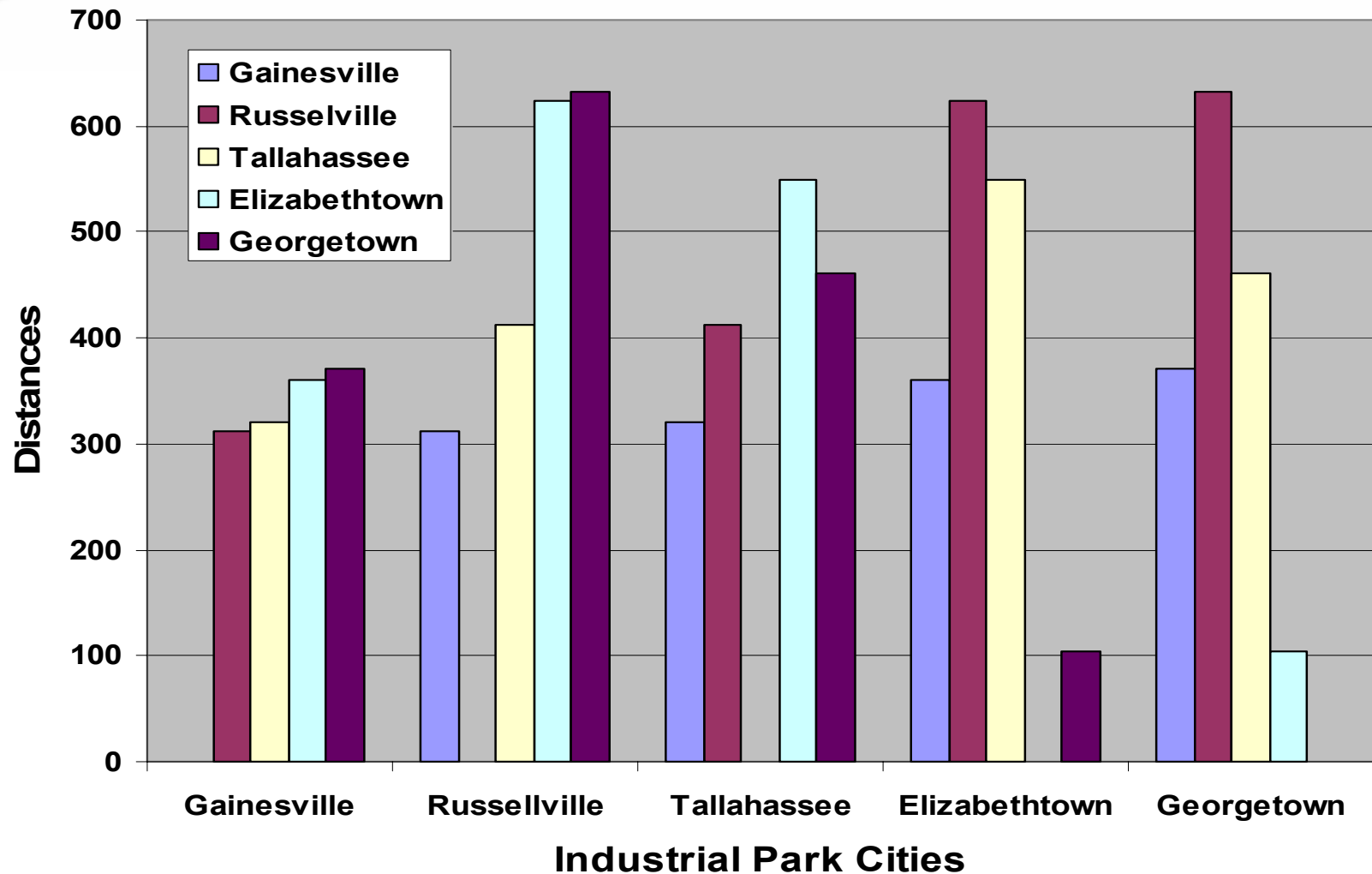
# Industrial Park Distance from Airport





# Industrial Park Cities Distance Range

Industrial Park Histogram







# En Route Simulation Facts

## – Decision Modules

- Weather Condition
- Maintenance
- Flying Time
- Delivery Time
- Available UAV (5, 10, 15)
- Service Requests

## – Inputs

- Number of Service Requests
- Distance
  - Original city
  - Destination city

## – Outputs

- Number of success/failure
- Number of services/day



# Critical Sim. Parameters

- Weight of the Plane
- Stall Speed
- Landing Velocity
- Wing Area (S)
- Horizontal Distance (X)
- Vertical Distance Height (Y)
- Air Density ( $\rho$ )
- Lift Coefficient
- Lift
- Drag Coefficient
- Drag
- Time to Descend (Min)
- Descend Delay Time
- Height Determination
- Distance to Runway
- Distance of Flare
- Flare Descend Time
- Ground Roll
- Stopping Distance (after ground roll)



# Landing Sequence

Plane will descend in a glide slope of 3 degree with WAAS always active.

**WAAS** – Center the Runway with Plane.

**INS** and **EGPWS** activate to find the runway distance and center it for successful landing.

**Vision** and **Radar Altimeter** activate to calculate the vertical distance from runway.

**Enhanced Ground Proximity Warning System (EGPWS)** – Situation of the Environments.

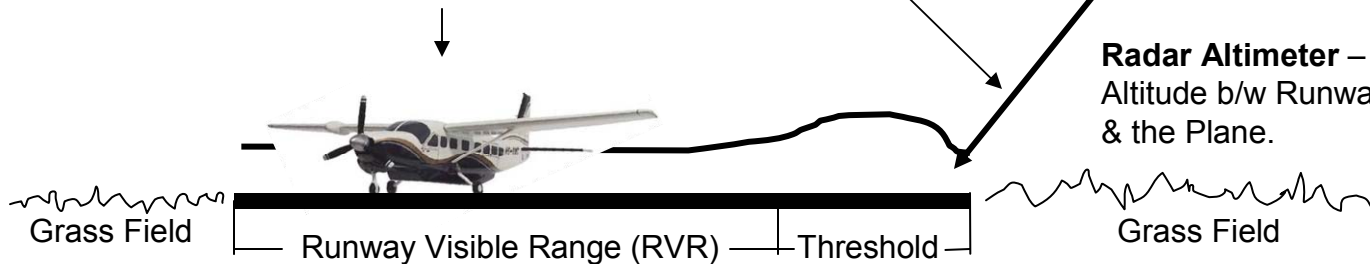
**Use of Flight Control Unit (GuideStar)** the plane is able to perform a flare maneuver.

**Optical System (Vision)** – Eye for the Plane.

**Ground Roll (Stopping Distance)** is calculated.

**Flight Control Unit (GuideStar)** – Brain for the Plane.

**Radar Altimeter** – Altitude b/w Runway & the Plane.



**Flare Height of 30 ft. with 75 knots of Speed to do a Successful Flare Maneuver.**



# Landing Parameters

- No Taxiing & Sequencing in Simulation/Analysis Plan
- Statistical Distributions for the variables are given by the tolerance range of overall components
- Weather conditions will be simulated by having a reduced friction coefficient
- Crosswind does not exceed 20% of stall velocity
- Landing velocity cannot be less than  $(1.3 * \text{stall speed})$
- Hot Temperature and Humidity will affect the ground roll